

# *Towards operational use of aircraft-derived observations: a case study at London Heathrow airport*

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# Towards operational use of aircraft-derived observations: a case study at London Heathrow airport.

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## 1 **Abstract**

2 Mode-Selective Enhanced Surveillance (Mode-S EHS) aircraft reports can be collected at a  
3 low-cost, and are readily available around busy airports. The new work presented here demon-  
4 strates that observations derived from Mode-S EHS reports can be used to study the evolution  
5 of temperature inversions since the data have a high spatial and temporal frequency. This  
6 is illustrated by a case study centred around London Heathrow airport for the period 4 to 5  
7 January 2015. Using Mode-S EHS reports from multiple aircraft and after applying quality  
8 control criteria, vertical temperature profiles are constructed by aggregating these reports at  
9 discrete intervals between the surface and 3000 m. To improve these derived temperatures,  
10 four smoothing methods using low-pass filters are evaluated. The effect of smoothing reduces  
11 the variance in the aircraft derived temperature by approximately half. After smoothing, the  
12 temperature variance between the altitudes 3000 m and 1000 m is 1 K to 2 K; and below 1000 m  
13 it is 2 K to 4 K. While the differences between the four smoothing methods are small, expo-  
14 nential smoothing is favoured because it uses all available Mode-S EHS reports. The resulting  
15 vertical profiles may be useful in operational meteorology for identifying elevated temperature  
16 inversions above 1000 m. However, below 1000 m they are less useful because of the reduced  
17 precision of the reported Mach number. A better source of in situ temperature observations

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18 would be for aircraft to use the meteorological reporting function of their automatic dependent  
19 surveillance (ADS) system.

## 20 **1 Introduction**

21 Weather impacts on airports are an important problem for society (Ball *et al.*, 2007; Markovic  
22 *et al.*, 2008; Barnhart *et al.*, 2012). In particular, fog and low visibility conditions reduce the  
23 air-traffic flow rates at airports as aircraft separations need to be increased to maintain safe  
24 operations. The reduced flow rate increases costs in terms of the extra fuel that must be used,  
25 loss of revenue due to reduced capacity at airports, environmental impacts on local air qual-  
26 ity and noise emissions, and climate impacts due to increased emissions of nitrogen oxides  
27 and carbon dioxide (Mahashabde *et al.*, 2011). Numerical weather prediction (NWP) forecast-  
28 ing fog and low visibility conditions is difficult since these require an accurate representation  
29 of orography, surface, boundary-layer fluxes and inversions in the vertical temperature profile  
30 (Stull, 2000; Jacobs *et al.*, 2008). Operational forecasting of temperature inversions depends  
31 on the availability of suitable observations (Roach *et al.*, 1976; Jacobs *et al.*, 2005; Fowler  
32 *et al.*, 2011) to locate the inversion. For example high-frequency reporting of vertical pro-  
33 files of temperature and wind may provide extra information for use in NWP assimilation and  
34 nowcasting (Dance, 2004; Rennie *et al.*, 2011; Simonin *et al.*, 2014; Sun *et al.*, 2014; Ballard  
35 *et al.*, 2015; James and Benjamin, 2017). Furthermore several authors (de Haan and Stoffelen,  
36 2012; de Haan, 2013; Strajnar *et al.*, 2015; Lange and Janjic, 2016) have demonstrated positive  
37 impacts in regional NWP models when assimilating derived observations from aircraft reports  
38 using Mode-Selective (Mode-S) Enhanced Surveillance (EHS), a system which transmits bi-  
39 nary coded messages to an aircraft's transponder and receives binary coded replies (Boisvert  
40 and Orlando, 1993; ICAO, 2010).

41 Strajnar *et al.* (2015, figure 7) showed that meteorological routine air reports (MRAR) of ambi-  
42 ent temperature, obtained from the secondary surveillance radar (SSR) using Mode-S, centred  
43 around Ljubljana airport, Slovenia, have a spatial and temporal resolution sufficient to locate a  
44 temperature inversion at around 1000 m above the surface. However, direct reports of ambient  
45 temperature using Mode-S MRAR is not routinely available since not all SSRs and not all air-  
46 craft are configured to make such reports. De Haan (2011) showed that Mode-S EHS reports  
47 of Mach number and true-airspeed, centred around Schipol airport, Netherlands, could be used  
48 to derive ambient temperature. In de Haan (2011, Figure 7) we noted that, after quality control  
49 and smoothing, the derived ambient temperature from a single aircraft profile may also locate  
50 temperature inversions. However, de Haan (2011); Mirza *et al.* (2016); Mirza (2017, table 6.2)

51 and [Stone \(2017\)](#) suggest that the uncertainty in the derived temperature from a single aircraft  
52 at low levels can range between 2 K and 10 K. This degree of uncertainty makes it difficult to  
53 locate the height and magnitude of the temperature inversion.

54 [Stone and Kitchen \(2015\)](#) showed that a mean temperature for a layer of thickness 2000 m  
55 could be computed using the global navigation satellite system's altitude reported by an air-  
56 craft's automatic dependent surveillance-broadcast (ADS-B) system. However, this method  
57 for determining thickness temperature is too coarse to resolve a temperature inversion.

58 All these methods use Mode-S/ADS-B reports from single aircraft to obtain temperature ob-  
59 servations. In our new work, we investigate the usefulness of using all available Mode-S EHS  
60 reports from multiple aircraft to estimate a vertical temperature profile.

61 In section 2 the current methods for obtaining in situ temperature measurements are described.  
62 Section 3 describes the method used to collect Mode-S EHS reports, how the Mach temperature  
63 observation is derived, and how these are aggregated to form a mean temperature observation.  
64 Section 4 defines four smoothing filters used to reduce the variance in Mode-S EHS reports.  
65 These are centred moving average, block average, linear regression and irregular exponential  
66 smoothing. In section 5 we apply the method described in section 3 to a case study based  
67 around London Heathrow to indicate the presence of temperature inversions. In section 6 we  
68 apply the four low-pass filters, to a sample of the data for the London Heathrow domain. In  
69 section 7 we show that the aggregated mean temperature profiles may provide useful informa-  
70 tion for operational meteorology, at least until temperature reports by ADS-B become more  
71 routinely available ([RTCA, 2012](#)). All times are expressed as Universal Time Coordinated  
72 (UTC).

## 73 **2 In situ Upper Air Temperature Observations.**

74 In situ observations of upper air temperature are made using a temperature sensor fixed to  
75 a device which ascends or descends between the surface and the top of the troposphere or  
76 beyond. Two types of such devices are the radiosonde and commercial aircraft.

77 For operational meteorology, modern radiosondes sample the atmosphere every second during  
78 ascent ([World Meteorological Organisation, 2014](#), Ch 12, p.348), which can take up to two  
79 hours. Typically, radiosondes are launched from fixed sites that are widely separated (approx-  
80 imately 100 km) and report at fixed times (usually 0000 and 1200 UTC) so do not provide  
81 sufficient horizontal spatial or temporal resolution to capture the onset or duration of a temper-  
82 ature inversion ([Fowler, 2010](#)).

83 The common method of receiving observations from commercial aircraft is from the Aircraft  
84 Meteorological Data Relay (AMDAR) program. An AMDAR equipped aircraft reports the  
85 horizontal wind and ambient temperature obtained from the aircraft's flight management sys-  
86 tem (Painting, 2003). These reports are compiled on-board the aircraft and are transmitted to  
87 a ground station. The frequency of transmission depends on the phase of flight (and whether  
88 the aircraft is configured to send a report). For example, an aircraft may be configured to re-  
89 port every 6 seconds for the first 90 seconds during ascent then every 20 seconds until level  
90 flight; during level-flight reports are every 3 to 10 minutes; during descent reports are every  
91 60 seconds (Painting, 2003, p.32).

92 In Europe, the AMDAR program is managed by E-AMDAR which provides at least one ver-  
93 tical profile once every three hours to participating National Meteorological Services (NMS)  
94 from around 100 airports across Europe. The Met Office obtains one vertical profile once ev-  
95 ery hour at major airports. In Europe and the UK, the reporting frequency of vertical profiles  
96 depends on the financial resources made available by the NMS. This contrasts with Air Traffic  
97 Management (ATM) which can interrogate an aircraft's transponder at a much higher frequency  
98 from a ground station SSR.

### 99 **3 Aggregation of Mode-S EHS Reports.**

100 Mode-S EHS is used by ATM to retrieve routine reports on an aircraft's state vector at a high  
101 temporal frequency (every 4 to 12 seconds). The aircraft's state vector consists of true-airspeed  
102 (hereafter referred to as the airspeed), magnetic-heading, ground speed, ground heading, al-  
103 titude and Mach number. These Mode-S EHS reports can be used to derive estimates of the  
104 ambient air temperature and horizontal wind at the aircraft's location (de Haan, 2011).

105 During the study period, the Met Office used a Mode-S EHS receiver network which consists  
106 of five receivers (Stone and Pearce, 2016). Reports that are actively polled for by ATM and  
107 those routinely broadcast by aircraft are collected and processed by the Met Office receiver  
108 network.

109 The Met Office Mode-S EHS receivers are co-located at sites used for the weather radar net-  
110 work, which provide a good line of sight of aircraft flying above 500 m, power supply and  
111 communication network. The Mode-S EHS reports are collated then transmitted in batches  
112 every 10 minutes to a central processing facility, where the data are then passed through a qual-  
113 ity control process (Stone and Pearce, 2016; Mirza, 2017). However, this network of Mode-S  
114 EHS receivers may be sub-optimal for the acquisition of Mode-S EHS reports at low levels,

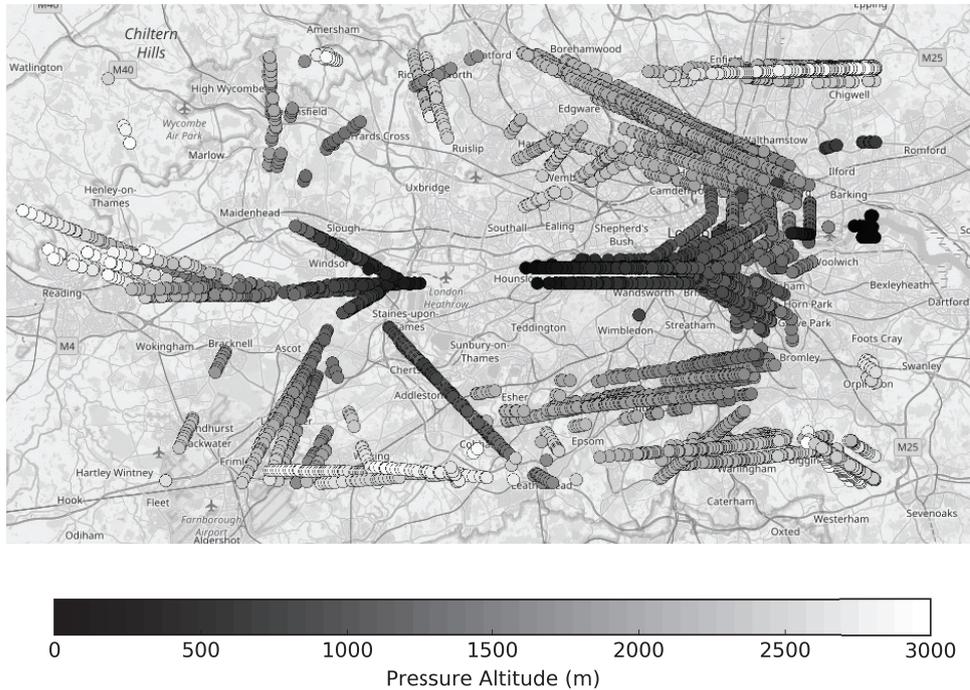


Figure 1: Spatial distribution of trajectories (circles, colour-coded by altitude) for ascending and descending aircraft within the London Heathrow domain, derived from Mode-S EHS reports received between 1200 to 1300 on 4 January 2015. The domain extends for a distance of 80 km east-west, 40 km north-south, height 3000 m from the surface, with London Heathrow airport at the domain's centre. Points where the aircraft's roll angle is greater than  $5^\circ$ , i.e. when turning, are removed since these data are considered unreliable. (Cartography ©OpenStreetMap contributors, licensed as CC BY-SA <https://www.openstreetmap.org/copyright>, 2018)

115 e.g., below 500 m, due to loss of the line of sight required to receive Mode-S EHS reports.

116 Figures 1 and 8 (see supplementary section) show the distribution of the Mode-S EHS reports  
 117 received from the Met Office Mode-S EHS receivers for a domain centred around London  
 118 Heathrow airport. The domain's dimensions are sufficient to contain the trajectories of aircraft  
 119 arriving at or departing from London Heathrow. Trajectories for descending aircraft are longer  
 120 than for ascending aircraft. The domain excludes the areas where aircraft are held prior to  
 121 their descent. The domain is not cuboid but can be imagined as an inverted truncated pyramid,  
 122 centred at the airport. (In the supplementary section, figure 9 shows the distribution of Mode-S  
 123 EHS reports for a domain centred around London Gatwick airport.)

124 The Mach Temperature,  $T_{MACH}$ , is derived from Mode-S EHS reports of Mach number,  $M$   
125 and airspeed,  $V_A$ , (de Haan, 2011; Mirza *et al.*, 2016), such that

$$T_{MACH} = \frac{T_0}{A_0^2} \left[ \frac{V_A}{M} \right]^2, \quad (1)$$

126 where the speed-of-sound  $A_0 = 340.294 \text{ ms}^{-1}$  and the assumed surface temperature  $T_0 =$   
127 288.15 K, are reference values defined at mean-sea-level pressure under international standard  
128 atmosphere conditions (ICAO, 1993).

129 To use as many of the Mode-S EHS reported data as possible they are aggregated to form a  
130 mean Mach Temperature,  $\bar{T}_{MACH}$ , observation. This ‘aggregated observation’ (Mirza *et al.*,  
131 2016; Mirza, 2017, Ch3) is the arithmetic mean of all the Mach Temperatures, derived using  
132 equation (1), for all Mode-S EHS reports received within a defined time period and in a spec-  
133 ified horizontal layer. The assigned position of  $\bar{T}_{MACH}$  is set at the centre of the horizontal  
134 layer and at the mean pressure altitude of all the reporting aircraft within. These layers form  
135 a vertical profile of  $\bar{T}_{MACH}$  observations when stacked in the vertical, which is centred around  
136 an airport.

137 We treat the errors as random so that the aggregated observation has a smaller error than an  
138 individual observation, since if the errors are random and uncorrelated then the standard error  
139 of the mean scales by  $1/\sqrt{n}$ , where  $n$  is the number of reports (Hoel, 1984, Ch 5 and Ch 10).

## 140 4 Temporal smoothing using low-pass filters

141 Studies by de Haan (2011); Mirza *et al.* (2016) have shown that Mach number and airspeed in  
142 equation (1) are subject to fluctuations which result in unrealistic values of derived tempera-  
143 ture. These fluctuations are thought to arise as a result of the reduced precision of these data  
144 caused by the Mode-S EHS transponder processing the data prior to its transmission. De Haan  
145 (2011) showed that by applying a linear smoothing algorithm to the time series of Mode-S  
146 EHS reported Mach number and airspeed of a single aircraft before computing the derived  
147 Mach Temperature then the large fluctuations in the latter are reduced. This action of linear  
148 smoothing is similar to that of a low-pass filter, which reduces high-frequency components of  
149 a time-varying signal. We apply and evaluate a selection of low-pass filters.

150 The low-pass filters described in this section are applied to the time series of Mode-S EHS  
151 reports for each aircraft trajectory and the result of the low-pass filter is used to generate a

152 new aircraft report. Using this filtered time series of reports the Mach Temperature report is  
 153 recomputed.

154 In our description of the filters, we use the notation  $x_k$ , for the value of an individual Mode-S  
 155 EHS report, with assigned time  $t_k$ . The filtered reports,  $X_{\bar{t}}$ , are computed by averaging over a  
 156 validation window of length  $W_L$ , and they are assigned a validity time,  $\bar{t}$ .

#### 157 4.1 Block-window average (BLK)

The block-window average method creates a time series of Mode-S EHS reports using the average of all reports within a validation window, of length  $W_L$ . The time series is split into a sequence of non-overlapping blocks then the average of each block is computed. In computing the average no report is used more than once. The newly filtered time series is given by,

$$X_{\bar{t}} = \frac{1}{2m+1} \sum_{j=-m}^{+m} x_{k+j} \quad \text{for } k = m+1, 3m+2, 5m+3, \dots, \left\lfloor \frac{N}{2m+1} \right\rfloor (2m+1) - m, \quad (2)$$

where  $N$  is the total number of reports in the time series and  $\left\lfloor \frac{N}{2m+1} \right\rfloor$  is the number of validation windows of length  $W_L = 2m+1$  in the dataset. (The floor operator  $\lfloor z \rfloor$ , gives the greatest integer that is less than or equal to  $z$  (Oldham *et al.*, 2010, p.68).) The validity time,  $\bar{t}$ , is given by,

$$\bar{t} = \frac{1}{2m+1} \sum_{j=-m}^{+m} t_{k+j}. \quad (3)$$

158 This method is simple to implement but is not robust. It is susceptible to large variations since  
 159 all the reports within the validation window are equally weighted.

#### 160 4.2 Centred moving average (CMA)

161 This is a straightforward method of computing a value over a short window length,  $W_L =$   
 162  $2m+1$ . This method is also known by other names, e.g., running-mean, running-average,  
 163 sliding-window average. Our method uses  $m$  reports before and after the current report, which  
 164 is at the centre of the window. Each report is weighted equally, so reports from the start to  
 165 the end of the window are treated to be of the same importance (Savitzky and Golay, 1964;  
 166 Wendisch and Brenguier, 2013). The new time series is given by,

$$X_{\bar{t}} = \frac{1}{2m+1} \sum_{j=-m}^{+m} x_{k+j} \quad \text{for } k = m+1, m+2, m+3, \dots, N-m, \quad (4)$$

167 with the validity time given by eq. (3).

168 However, this method is also not robust since it can be affected by large outliers, and fluctua-  
169 tions in the new time series may lag behind those seen in the original time series, although the  
170 magnitude of the variations is reduced.

### 171 4.3 Piece-wise linear regression (LIN)

This uses the least squares regression method to compute a local rate of change, which is assumed to be linear over the validation window,  $W_L$ . In other words, the mean values obtained from fitting a straight line to the data locally are used to create the new time series. This is a statistical method that minimises the differences between a control variable and predicted values. The new time series is given by

$$X_{\bar{t}} = \alpha \bar{t} + \beta, \quad (5)$$

where the validity time is given by eq. (3). The local constant,  $\beta$ , is defined as

$$\beta = \bar{x} - \alpha \bar{t}. \quad (6)$$

where

$$\bar{x} = \frac{1}{2m+1} \sum_{j=-m}^{+m} x_{k+j}, \text{ for } k = m+1, m+2, m+3, \dots, N-m, \quad (7)$$

i.e., the local mean  $\bar{x}$  computed over the window. The corresponding local rate of change,  $\alpha$ , (i.e., the gradient) is given by,

$$\alpha = \frac{\sum_{j=-m}^{+m} (x_{k+j} - \bar{x})(t_{k+j} - \bar{t})}{\sum_{j=-m}^{+m} (t_{k+j} - \bar{t})^2}. \quad (8)$$

172 Unlike the centred moving average this method is more responsive to variations in the time  
173 series.

### 174 4.4 Irregular exponential moving average (IRR)

175 The exponential smoothing method is similar to the centred moving average except observa-  
176 tions are weighted according to their position in time. The current observation is weighted more  
177 than the observations made at earlier times. The simple exponential moving average (Brown,  
178 2004; Kim and Huh, 2011) assumes observations are available at regular time intervals. How-

179 ever, since the Mode-S EHS reports used to construct aircraft trajectories may be at irregular  
 180 time intervals and there may be missing data, the [Wright \(1986\)](#) method is used, which extends  
 181 the exponential smoothing method to irregular time intervals. The new time series is given by,

$$X_{t_k} = (1 - V_k)X_{t_{k-1}} + V_k x_{t_k}, \quad (9)$$

where

$$V_k = \frac{V_{k-1}}{b_k + V_{k-1}} \quad (10)$$

and

$$b_k = (1 - a)^{(t_k - t_{k-1})}, \quad (11)$$

182 for  $k = 2, 3, 4, \dots, N$ , and  $0 \leq a < 1$ .

183 The value  $a$  is a smoothing parameter which determines the proportion of the new information  
 184 to be added to the running average. The parameter  $V_k$  is a weighting function which is given  
 185 an initial value of  $V_1 = 1$ . The larger the value of the parameter  $V_k$ , the less weight is given to  
 186 the running average. The weighting function depends on the time separation between reports.  
 187 For each  $X_{t_k}$  the assigned validity time is  $t_k$  since the former directly replaces each  $x_{t_k}$ .

#### 188 **4.5 Consistency check**

189 We apply a consistency check so that the horizontal spatial and temporal resolutions of the time  
 190 series are reasonably consistent along the aircraft trajectory. This consistency check is applied  
 191 because there are fewer Mode-S EHS reports along an aircraft's trajectory than are actually  
 192 available in principle.

193 We assume that a break in the time series of reports arise as a result of either (a) the aircraft  
 194 exiting from a turning point on its approach to land, (b) that it passed out of then re-entered  
 195 the airport domain, shown in figure 1, (c) that the aircraft was not within the line of sight  
 196 reception to the Mode-S EHS receiver or (d) due to quality control pre-processing of Mode-S  
 197 EHS reports, performed at the monitoring site ([Stone and Pearce, 2016](#)), which removes reports  
 198 when an aircraft's roll angle exceeds 5 degrees creating gaps in the time series of reports.

199 The consistency check is used to determine when a low-pass filter outputs a filtered value.  
 200 The filtered value  $X_{\bar{t}}$  is set to a missing data indicator when the time difference between two  
 201 successive reports,  $\delta t$ , used to compute the filtered value is greater than a maximum permitted  
 202 time difference,  $\delta t > \delta t_{max}$ . (This affects the BLK low-pass filter more as reports are only  
 203 used once.) The value of  $\delta t_{max}$  ensures that the data input to the low-pass filter are closely

204 related in time and space.

205 We select a value for  $\delta t_{max}$  equal to the standard deviation of the time difference between  
206 successive Mode-S EHS reports along an aircraft’s trajectory. For the selected day we use all  
207 aircraft trajectories to compute this standard deviation. The result is rounded to the nearest  
208 whole second.

209 The effect of applying the consistency check is to set the maximum time window for sampling  
210 the meteorological conditions based on the validation window of length  $W_L$ .

## 211 5 Inversion Case Study

212 In this section, we use a case study to identify useful meteorological information for the London  
213 Heathrow domain between 4 and 5 January 2015. This period was chosen because fog was a  
214 persistent weather feature. One of the meteorological conditions for fog to arise is the presence  
215 of a temperature inversion at low altitude or near the surface.

### 216 5.1 Observations

217 To assess the information content of the  $\overline{T}_{MACH}$  vertical profile we compare it to temperature  
218 reports from other observation systems. We use the high-resolution temperature profile from  
219 Herstmonceux, the nearest radiosonde station. We also use AMDAR temperature reports. We  
220 note also that all AMDAR reporting aircraft also report Mode-S EHS. We assume that ra-  
221 diosonde and AMDAR observations are representative of the meteorological conditions. The  
222 vertical profile of  $\overline{T}_{MACH}$  is compared to the forecast mean vertical temperature profile from the  
223 Met Office’s limited-area, high-resolution, convection-permitting NWP model for the United  
224 Kingdom, the UKV ([Lean et al., 2008](#); [Tang et al., 2013](#)); the mean is calculated using UKV  
225 vertical profiles at selected points across the London Heathrow domain. We note that the ra-  
226 diosonde and AMDAR temperature reports that we use for comparison are not assimilated by  
227 the UKV.

228 In figure 2 we show all temperature reports for the London Heathrow domain on 4 January  
229 2015 with a validity time of 0600, that is all observations received between 0530 and 0630.  
230 The  $\overline{T}_{MACH}$  profile (black triangles) is constructed using the aggregation method described in  
231 section 3. The  $\overline{T}_{MACH}$  error bars (black) are the 95% confidence limits for the mean using  
232 the Student-t distribution ([Hoel, 1984](#), Ch 5 and Ch 11). For comparison, we show in situ ob-  
233 servations from two other observing systems: radiosonde and AMDAR. The radiosonde was  
234 launched at 0515, headed due south of its launch site at Herstmonceux and reached an altitude

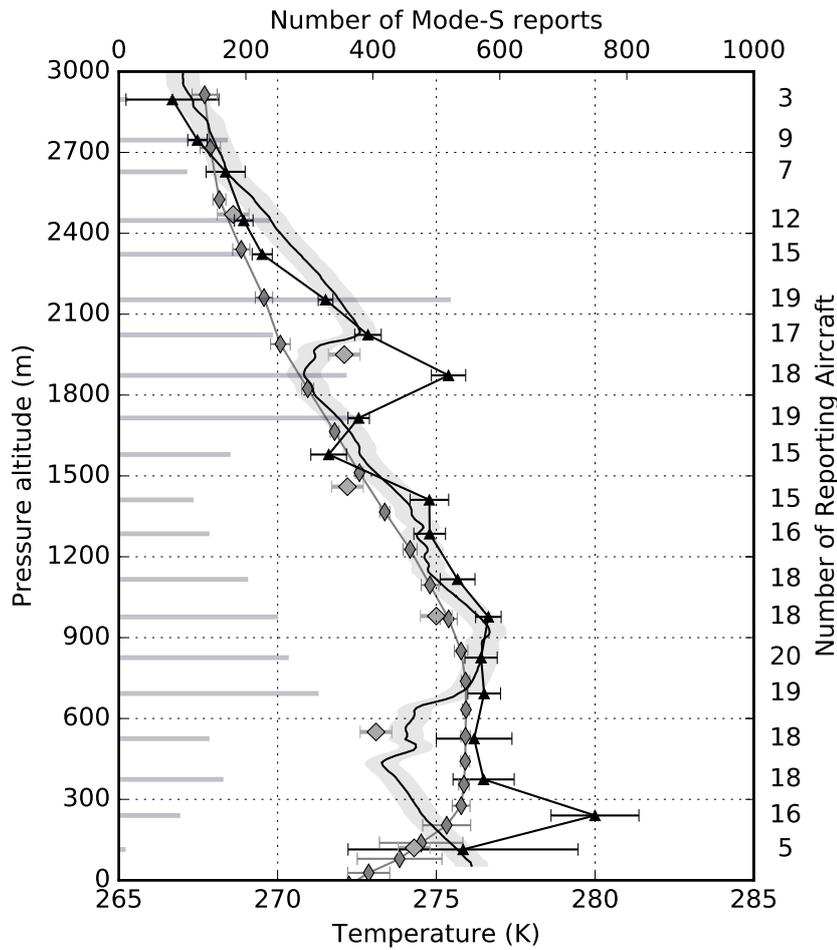


Figure 2: Temperature reports for the London Heathrow domain on 4 January 2015 for the period 0530 to 0630. Aggregated Mach Temperature observation and its 95% confidence interval (black triangles), with the number of Mode-S EHS reports used shown by the horizontal bars from the left, and the number of reporting aircraft shown on the right-axis. Herstmonceux radiosonde report valid at 0600 (black solid line, with its reported precision of  $\pm 0.5$  K shown by the grey shading), AMDAR reports (large diamonds) and their reported precision of  $\pm 0.5$  K (error bars), and the mean UKV forecast and its 95% confidence interval, valid at 0600 from the forecast run at 0300 4 January 2015 (narrow diamonds).

235 of 3000 m at 0524. Position and temperature reports were made every 2 s. The region of the  
 236 atmosphere sampled by the radiosonde is not contained within the London Heathrow domain.  
 237 AMDAR temperature reports are shown as point observations (Painting, 2003), received be-  
 238 tween 0557 and 0617 from an aircraft destined to land at London Heathrow during this period.  
 239 We also show the mean UKV forecast temperature profile for the London Heathrow domain

240 with the validity time 0600. The mean forecast temperature profile is computed by using a  
241 sample of nine 1-D column profiles from across the London Heathrow domain (Mirza, 2017,  
242 Fig 5.5). The standard deviation of the mean forecast temperature profile indicates that at this  
243 time between the pressure altitude range 300 m and 3000 m there is little variation across the  
244 domain ( $<0.5$  K) and below 300 m it is around 1.5 K. (For this pressure altitude range In-  
245 gleby and Edwards (2014) estimated that the average UKV model error to be  $\pm 0.75$  K when  
246 compared against high-resolution radiosonde reports.)

## 247 5.2 Observed Meteorological Features

248 In figure 2 the radiosonde report indicates the presence of two temperature inversions: a low-  
249 level temperature inversion between 500 m and 900 m, reported at 0516, and an elevated tem-  
250 perature inversion between 1800 m and 2000 m, reported at 0520. The AMDAR observations,  
251 reported between 0557 and 0612, are broadly in agreement with the radiosonde. These in situ  
252 observations provide a broad description of the vertical temperature structure of the atmosphere  
253 between Heathrow airport and Herstmonceux. However, there is a clear difference between  
254 these in situ observations and the mean UKV forecast for the London Heathrow domain.

255 The UKV at 0600 forecasts a low-level inversion between the surface and 300 m but does not  
256 forecast the elevated inversion between 1800 m and 2000 m. However, the  $\overline{T}_{\text{MACH}}$  observa-  
257 tions, obtained between 0530 and 0630, do suggest that an elevated inversion is present.

258 The radiosonde and AMDAR reports were not included in the UKV analysis (i.e., the initial  
259 state of the NWP model) as they were received after the data assimilation observation pro-  
260 cessing period, 0130 to 0419. Therefore the UKV forecast will not have taken into account  
261 the existence and the location of the temperature inversions shown by these observations and  
262 there are no other sources of in situ upper air temperature observations during the observation  
263 processing period. Furthermore, the elevated temperature inversion is not forecast by the UKV  
264 at 0300, 0400 and 0500 within the London Heathrow domain, but this may also be due to  
265 deficiencies in the physical modelling within the UKV.

266 The  $\overline{T}_{\text{MACH}}$  observations appear consistent with the radiosonde and AMDAR reports between  
267 700 m and 3000 m. In this case, while there are insufficient AMDAR reports to resolve the  
268 inversion, its presence is shown by the  $\overline{T}_{\text{MACH}}$  observations at around 1900 m, even though  
269 the magnitude of the inversion suggested by the  $\overline{T}_{\text{MACH}}$  report differs significantly from that  
270 shown by the radiosonde. The radiosonde and AMDAR show the inversion to be higher, but  
271 this difference could be accounted for by a horizontal variation in the inversion height. Below  
272 700 m the  $\overline{T}_{\text{MACH}}$  observations are more consistent with the UKV forecast, except around

273 300 m, where the difference between the UKV and  $\overline{T}_{\text{MACH}}$  is of the same magnitude as at  
274 2000 m, i.e., approximately 5 K.

275 The absence of the elevated temperature inversion at around 2000 m in the UKV forecast would  
276 be important for the subsequent forecasts of other meteorological phenomena. An elevated in-  
277 version in effect caps vertical movement and dispersion of atmospheric aerosols. This may  
278 affect the forecast conditions for solar insolation and the formation or persistence of fog and  
279 cloud (Fowler *et al.*, 2011). We suggest that  $\overline{T}_{\text{MACH}}$  observations could provide an additional  
280 source of information, albeit a qualitative source, on the vertical temperature profile that may  
281 otherwise be unknown, since the 0600 Herstmonceux radiosonde report is made only on de-  
282 mand (unlike the reports at 0000 and 1200). We illustrate the qualitative information contained  
283 in the  $\overline{T}_{\text{MACH}}$  observations in figure 3.

284 Figure 3 shows the temperature reports available for the validity time 0900 on 4 January 2015;  
285 these are all reports received between 0830 and 0930. There are no in situ observations from  
286 radiosonde because there is no routine launch at this time of day. The 13 AMDAR observations  
287 were reported between 0830 to 0837 from an aircraft on a descent path to Heathrow airport. The  
288 computation and depiction of the  $\overline{T}_{\text{MACH}}$  observations and UKV vertical temperature profile are  
289 as described in figure 2. We note that  $\overline{T}_{\text{MACH}}$  observations suggest that the elevated inversion  
290 noted in figure 2 still persists although at a lower altitude, between 1500 m and 1800 m, with  
291 a broadly isothermal region between 1000 m and 1500 m. The AMDAR reports are broadly in  
292 agreement with the presence of the temperature inversion but not with the isothermal region.  
293 The UKV forecast for these two regions does not show either meteorological feature. The  
294 AMDAR reports would not have been available for assimilation into the UKV. Figure 4 shows  
295 the same time period but 24 hours later for which there are no AMDAR or radiosonde reports.  
296 In this case, the UKV forecast and the  $\overline{T}_{\text{MACH}}$  observations show some agreement indicating  
297 the presence of an elevated temperature inversion between 1000 m and 1500 m. Thus, in the  
298 absence of other in-situ observations, the  $\overline{T}_{\text{MACH}}$  observations could provide useful information  
299 about the vertical structure of the atmospheric temperature.

300 Figures 2, 3 and 4 all show that  $\overline{T}_{\text{MACH}}$  indicates warmer conditions compared to the UKV  
301 forecast. This may be due to a bias in  $\overline{T}_{\text{MACH}}$  resulting from the numbers of aircraft that are  
302 ascending and descending at any given time (although it is also possible that the UKV NWP  
303 model is biased). Studies by Mirza (2017) and Stone (2017) suggest that  $\overline{T}_{\text{MACH}}$  reports be-  
304 tween the surface and 3000 m appear cooler than the ambient conditions when aircraft ascend,  
305 while for descents these reports appear warmer. These effects may be the result of aircraft  
306 manoeuvrings during ascent or descent. For example, most descending aircraft extend their

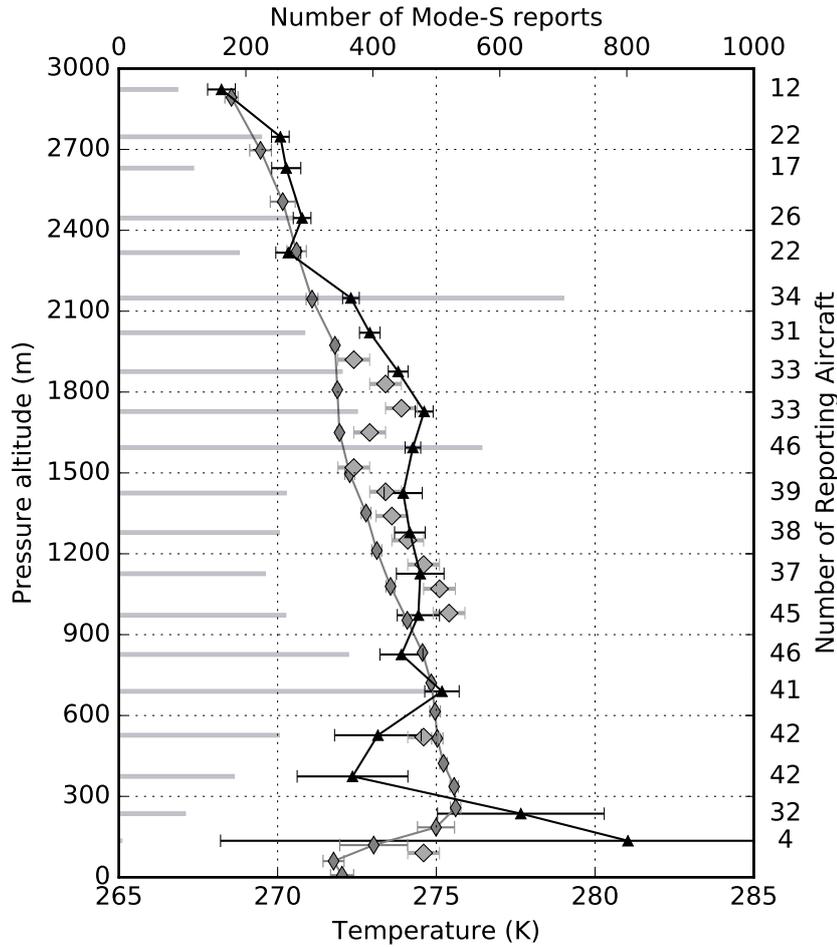


Figure 3: Temperature reports for the London Heathrow domain on 4 January 2015 for the period 0830 to 0930. Symbols are as described in figure 2. This plot shows the aggregated Mach Temperature reports and the corresponding number of Mode-S EHS reports, AMDAR reports, and the mean UKV forecast valid at 0900 .

307 landing gear and set full flaps at a height of around 300 m. This causes a strong deceleration,  
 308 which could explain major deviations of the reported Mach number from the observed airspeed  
 309 and thus erroneous temperatures. In addition, the height where the  $\bar{T}_{\text{MACH}}$  profile deviates from  
 310 the other data coincides with the bottom (and the most probably populated) level of London  
 311 Heathrow's holding patterns at 2000 m. Aircraft on hold do significantly more manoeuvring  
 312 which may lead to a decrease in the accuracy of the derived  $T_{\text{MACH}}$  reports. [Mirza et al. \(2016,](#)  
 313 [Figure 11\)](#) suggest that with sufficient Mode-S EHS reports from a single aircraft type, e.g.,  
 314 greater than 100 at each altitude interval, then any bias may be reduced to near zero. However,  
 315 [Stone \(2017, Figure 1b\)](#) suggests that the bias may depend on whether the aircraft is ascending

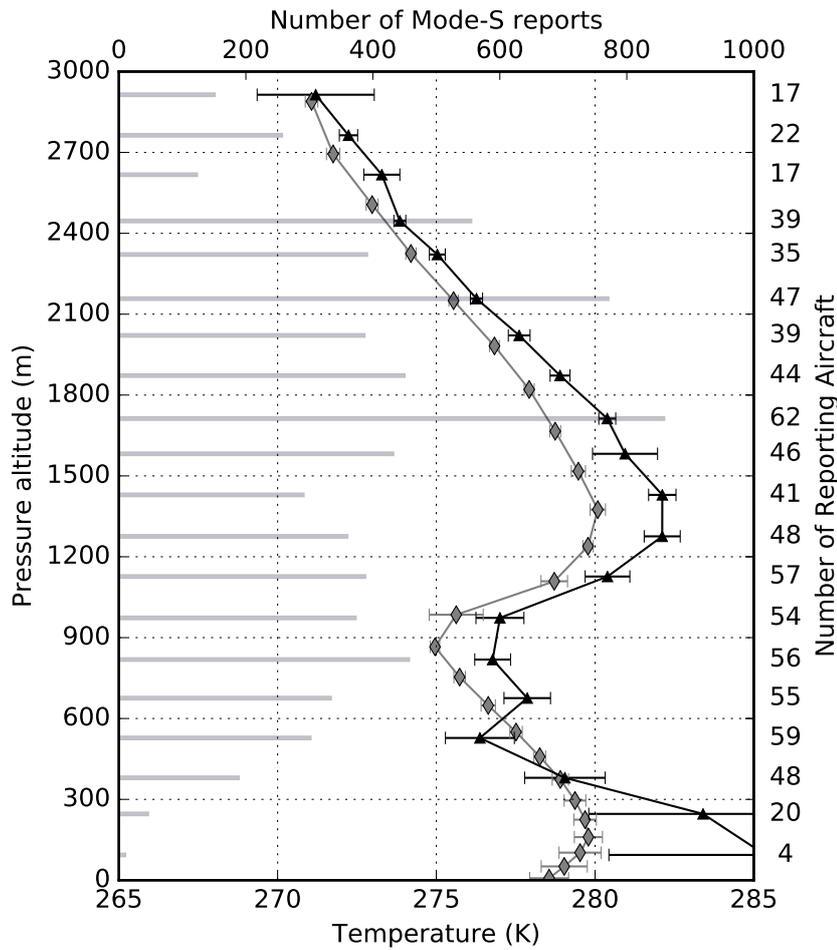


Figure 4: Temperature reports for the London Heathrow domain on 5 January 2015 for the period 0830 to 0930. Symbols are as described in figure 2. This plot shows the aggregated Mach Temperature reports and the corresponding number of Mode-S EHS reports and the mean UKV forecast valid at 0900. The lowest two points (not shown) are  $283.4 \pm 3.6$  K and  $285.3 \pm 4.9$  K. There were no radiosonde or AMDAR reports available for this time period and altitude range.

316 or descending. Further research is needed to understand these effects, for example. a much  
 317 longer study such as was done for AMDAR (Drue *et al.*, 2008).

318 Figure 5 shows similar temperature reports as shown in figure 4 but for the validity time at  
 319 2100 on 5 January 2015; these are all reports received between 2030 and 2130. There are no  
 320 radiosonde observations, but there were 9 AMDAR reports received between 2043 and 2045  
 321 from an aircraft departing from Heathrow. The UKV mean profile is for 2100 from the forecast  
 322 run at 2100 on 5 January 2015, so this represents the NWP analysis. Unlike the previous

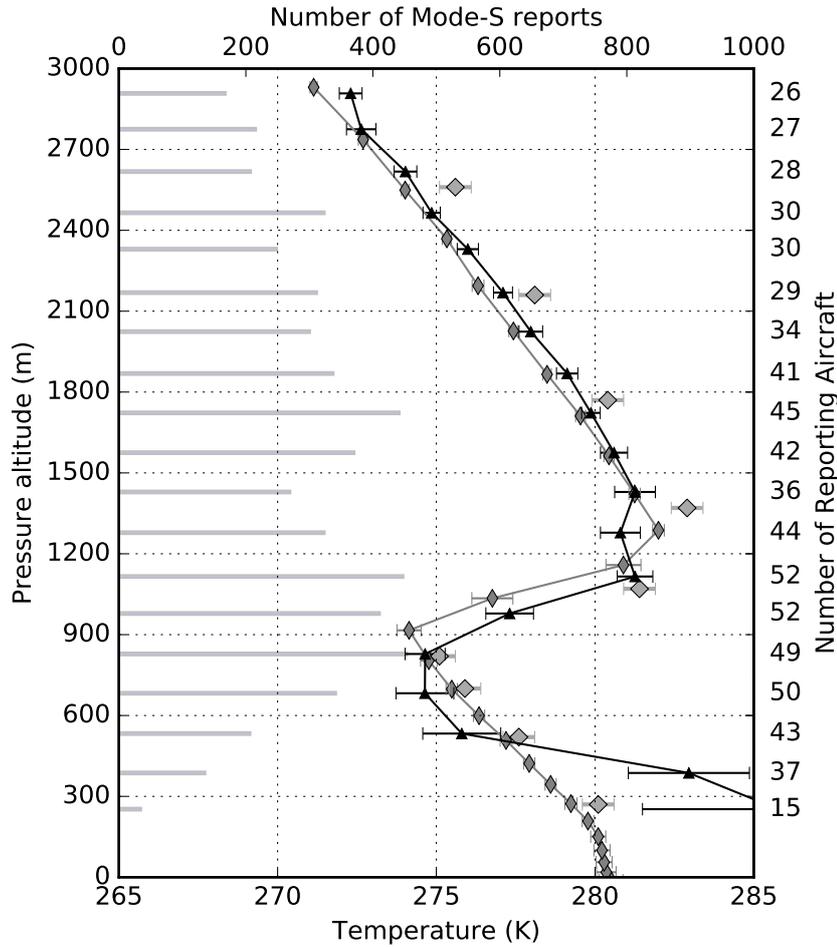


Figure 5: Temperature reports for the London Heathrow domain on 5 January 2015 for the period 2030 to 2130. Symbols are as described in figure 2. This plot shows the aggregated Mach Temperature reports and the corresponding number of Mode-S EHS reports, available AMDAR reports and the mean UKV forecast valid at 2100 from forecast run at 2100 on 5 January 2015. The lowest two points (not shown) are  $283.0 \pm 1.9$  K and  $285.8 \pm 4.3$  K. There were no radiosonde reports available for this time period and altitude range.

323 examples, it is likely that the AMDAR reports were received in time for their assimilation prior  
 324 to the UKV forecast run. Therefore there is a good correspondence between the AMDAR  
 325 temperature reports and UKV mean temperature profile. The  $\bar{T}_{\text{MACH}}$  observations between  
 326 600 m and 3000 m also show a good correspondence, in particular capturing the elevated  
 327 inversion between 900 m and 1500 m. However, in each of the cases shown, at or below  
 328 1000 m the  $\bar{T}_{\text{MACH}}$  observations show increased level of uncertainty, as shown by the 95%  
 329 confidence limits, and large differences between the AMDAR and radiosonde observations,

330 and the UKV forecasts.

331 The radiosonde and AMDAR reports are effectively instantaneous values, reporting on a time  
332 scale of seconds to minutes. The  $\overline{T}_{\text{MACH}}$  observation uses all available Mode-S EHS reports  
333 over a large spatial domain, and is an average over the hour thus representing the mean condi-  
334 tions in space and time. The horizontal bars shown in figure 2 indicate the number of observa-  
335 tions used to compute each  $\overline{T}_{\text{MACH}}$  observation. The mean time difference between reports is  
336 2 seconds per aircraft which corresponds to a horizontal spatial sampling scale around 250 m,  
337 however, any variability on this scale will be lost due to the averaging process. Where there is  
338 an agreement between the  $\overline{T}_{\text{MACH}}$  observation and the UKV this may be due to the latter also  
339 representing the mean conditions over the hour, although its spatial sampling scale is 1500 m.

340 We do note that  $\overline{T}_{\text{MACH}}$  observations show a degree of variability, as represented by the 95%  
341 confidence limits. The large variation in the computed  $\overline{T}_{\text{MACH}}$  observations may be due to the  
342 low precision of the underlying data, mainly the Mach number (de Haan, 2011; Mirza, 2017).  
343 The large uncertainty in the confidence limits is due to the drop in the available number of  
344 Mode-S EHS reports used to compute the  $\overline{T}_{\text{MACH}}$  observations. Using the Student- $t$  distribu-  
345 tion to compute the confidence limits may be unreliable or unsuitable at these low levels as  
346 the distribution of the individual  $T_{\text{MACH}}$  reports becomes multi-modal. Since the atmospheric  
347 conditions do not appear to vary greatly over the hour, we suggest that variability of  $\overline{T}_{\text{MACH}}$  ob-  
348 servations is likely to be due to the precision of the Mode-S EHS data used to derive the Mach  
349 Temperature (Mirza *et al.*, 2016; Mirza, 2017). This results in the poor characterisation of the  
350 vertical temperature profile at levels below 1000 m. (Figures 10 and 11 in the supplementary  
351 section (available online) show examples of the derived profiles for a similar size domain with  
352 London Gatwick airport at its centre for the same case study period.)

353 This variability is not seen in the radiosonde and AMDAR reports, especially at levels below  
354 1000 m. However, there are insufficient AMDAR reports to characterise fully the vertical  
355 temperature profile, and so they may not capture inversions between the surface and 600 m. The  
356 low reporting of AMDAR may be due to operational constraints, e.g., availability of suitably  
357 equipped aircraft or cost constraints which limit reporting to a single aircraft.

## 6 Temporal smoothing using low-pass filters

### 6.1 Motivation for low-pass filtering

In section 5.2 it was shown that  $T_{\text{MACH}}$  reports are subject to a high degree of variability especially at altitudes below 1000 m. De Haan (2011) and Mirza (2017) suggest the variability is due to the effects of Mode-S EHS processing. In this section, we apply four methods that perform the function of a low-pass filter, described in section 4, to a sample of the data for the London Heathrow domain. The filters are applied to the time series of Mode-S EHS reports for each aircraft within the London Heathrow domain. The filters create a new time series of smoothed Mode-S EHS reports which are then used to compute  $T_{\text{MACH}}$  observations (eq. (1)).

Figure 6 has four panels. Each panel shows the same short time series of non-smoothed Mode-S EHS reports (grey dots) for Mach number and airspeed, and over the period of one minute there are 28 reports of each. The corresponding derived Mach Temperature ranges between 269 K and 291 K. However, such a change in the ambient temperature in one minute is unrealistic. De Haan (2011) suggests that this magnitude of change in Mach Temperature is due to the low precision of the reported Mach number. Mirza (2017) shows that this is indeed the case but then goes on to suggest that the variation in Mach Temperature is also due to the asynchronous changes in the Mode-S EHS reports of Mach number and airspeed. Close examination of figure 6(a) shows the effects of low precision and asynchronous changes.

In figure 6(a)(i) the first six Mach number reports show there are two step changes of -0.004 while the airspeed remains constant, indicated by region A, figure 6(a)(ii). These step changes represent the reporting precision of the Mach number after Mode-S EHS processing. The corresponding Mach Temperature, figure 6(a)(iii), computed using equation (1), show step changes of +7 K. These changes occurred over 9 s with two step changes in altitude: 1821 m to 1814 m to 1806 m (not shown). Equation (1) suggests that if the airspeed is constant then a decrease in the Mach number corresponds to an increase in Mach Temperature. This is also suggested by figure 2 where for the altitude range of these Mode-S EHS reports the radiosonde and AMDAR reports indicate the presence of a temperature inversion.

In figure 6(a)(ii), the report at region B for airspeed shows a large step change of -8 knots while the Mach number and altitude are unchanged. This results in a step change of -21 K in the corresponding Mach Temperature in 1 s. Equation (1) suggests that if the Mach number is constant then a decrease in airspeed corresponds to a decrease in the Mach Temperature. However, for the 1 s over which this change takes place the aircraft's reported altitude remained at 1806 m and its horizontal displacement was 138 m. It is unlikely that the actual ambient

391 temperature would change by this magnitude over such a short distance and time. But if we  
392 assume that temperature is constant then equation (1) shows that a decrease in airspeed should  
393 show a corresponding decrease in Mach number, which in this instance did not occur. We  
394 suggest, therefore, that the Mode-S EHS processing causes asynchronous changes in the Mach  
395 number and airspeed which may result in the observed large fluctuations in Mach Temperature.

396 Regions C and D show a synchronous change in Mach number and airspeed, which results in  
397 a change of Mach Temperature of -9.5 K. The changes in altitude for each occurrence were  
398 1783 m to 1768 m over 5 s and 1737 m to 1722 m over 4 s. We suggest that the change in  
399 magnitude, while smaller than for the asynchronous case at region B, is due to the Mode-S  
400 EHS processing which reduces the precision of the Mach number and airspeed.

401 In summary, there are two effects of Mode-S EHS processing that may account for the observed  
402 variability in the derived Mach Temperature: the reduced precision of the reported Mach num-  
403 ber and airspeed and their asynchronous changes. The use of a suitable low-pass filter may  
404 smooth out the step changes in Mach number and airspeed thus reducing the observed vari-  
405 ability in the derived Mach Temperature. We consider the use of low-pass filters in the next  
406 section.

## 407 **6.2 Applying low-pass filters to time series of Mode-S EHS Reports**

408 We now explain how we set-up and use the low-pass filters. For the London Heathrow domain,  
409 the consistency check  $\delta t_{max}$  is 6 s. For BLK (eq. (2)), CMA (eq. (4)) and LIN filters (eq.  
410 (5)) the validation window is set with  $m = 2$ . This provides five reports for the validation  
411 window, i.e., where each filtered report has two reports either side, which are used to compute  
412 the mean value, except at the start and end of the time series. If 6 s is the maximum time  
413 separation between the five reports within the validity window then the filtered report represents  
414 the meteorological conditions sampled over 30 s. This is an appropriate sample time given  
415 that aircraft are changing position horizontally and vertically. Typical ascent rates are 5-10  
416  $\text{ms}^{-1}$  so a 30 s averaging could be over 150-300 m in the vertical. This is similar to the  
417 vertical grid length in many NWP models. Typical glide speed would be 100-120  $\text{ms}^{-1}$  giving  
418 a horizontal representation over 3.0-3.6 km. During the sampling time the aircraft may make  
419 control movements that increases or decreases its altitude during any part of its phase of flight:  
420 ascent, en-route or descent. These may be considered as an additional source of high-frequency  
421 noise.

422 There is a trade-off between the parameters  $\delta t_{max}$  and  $m$ . If  $\delta t_{max}$  is too short in time then  
423 high-frequency components may not be sufficiently damped. Furthermore, this limits the num-

424 ber of reports used due to failing the consistency check (see section 4.5). If the window length is  
 425 too large then over-smoothing may result which may cause the position and altitude of the tem-  
 426 perature inversion to be either misplaced or not detected. However, these parameters could be  
 427 tuned for particular operational conditions at airports or to apply different consistency checks  
 428 for ascending and descending aircraft since rates of ascent are larger than rates of descent. The  
 429 additional outputs of these low-pass filters (except IRR) are the means of the time, latitude,  
 430 longitude and pressure altitude quantities within the validation window.

431 For IRR (eq. (9)), we use a smoothing factor  $a = 0.2$ . The weighting function (eq. (11)) is  
 432 initialised with the time difference  $t_k - t_{k-1} = 1$  s. These parameters were selected so that when  
 433 the time separation between reports is 4 s, the expected SSR rotation rate, then the exponential  
 434 smoothing will weight the previous filter value and the current observation equally. Thus the  
 435 IRR low-pass filter replaces each Mode-S EHS report in the aircraft's trajectory, therefore, the  
 436 low-pass-filtered trajectory contains the same number of reports.

### 437 6.3 Effect of applying low-pass filters

438 In figure 6 the resulting smoothed Mach number, airspeed and recomputed Mach temperature  
 439 are shown as the square points after applying the low-pass-filters discussed in section 4. The  
 440 main effect of the low-pass filters IRR, CMA, and LIN (figures 6(b), 6(c), 6(d) respectively) is  
 441 to smooth the step transitions in Mach number and airspeed which reduces the variance of the  
 442 Mach Temperature distributions at each altitude bin. This is the desired effect as it shows that  
 443 the impact of the high-frequency components is being diminished.

444 We apply each of these low-pass filter methods to all aircraft trajectories within the London  
 445 Heathrow domain. We then apply the aggregation method to recompute  $\bar{T}_{\text{MACH}}$  for each hor-  
 446 izontal layer (shown in figure 2). Figure 7(a)(i) shows the results after applying the different  
 447 low-pass filters. Figure 7(a)(ii) shows the difference between the smoothed and unsmoothed  
 448  $\bar{T}_{\text{MACH}}$  observations. Above 1000 m the difference ranges between  $\pm 0.5$  K. However, below  
 449 1000 m the magnitude of the smoothed  $\bar{T}_{\text{MACH}}$  is greater. The magnitude of the latter results  
 450 may arise because reports have been filtered out during the low-pass filtering. This is shown in  
 451 figure 7(b)(ii) where the number of reports for CMA and LIN are less than for IRR (the number  
 452 of reports for the unsmoothed profile is the same as for the IRR). The number of reports for  
 453 BLK low-pass filter is greatly reduced but this is expected since this method replaces a series of  
 454 reports with a single report whereas the other low-pass methods use substitution. The overall  
 455 effect of the applying the low-pass filters to the computed  $\bar{T}_{\text{MACH}}$  is minimal. However, the  
 456 low-pass filters have a greater effect on the computed standard deviation of the  $\bar{T}_{\text{MACH}}$ .

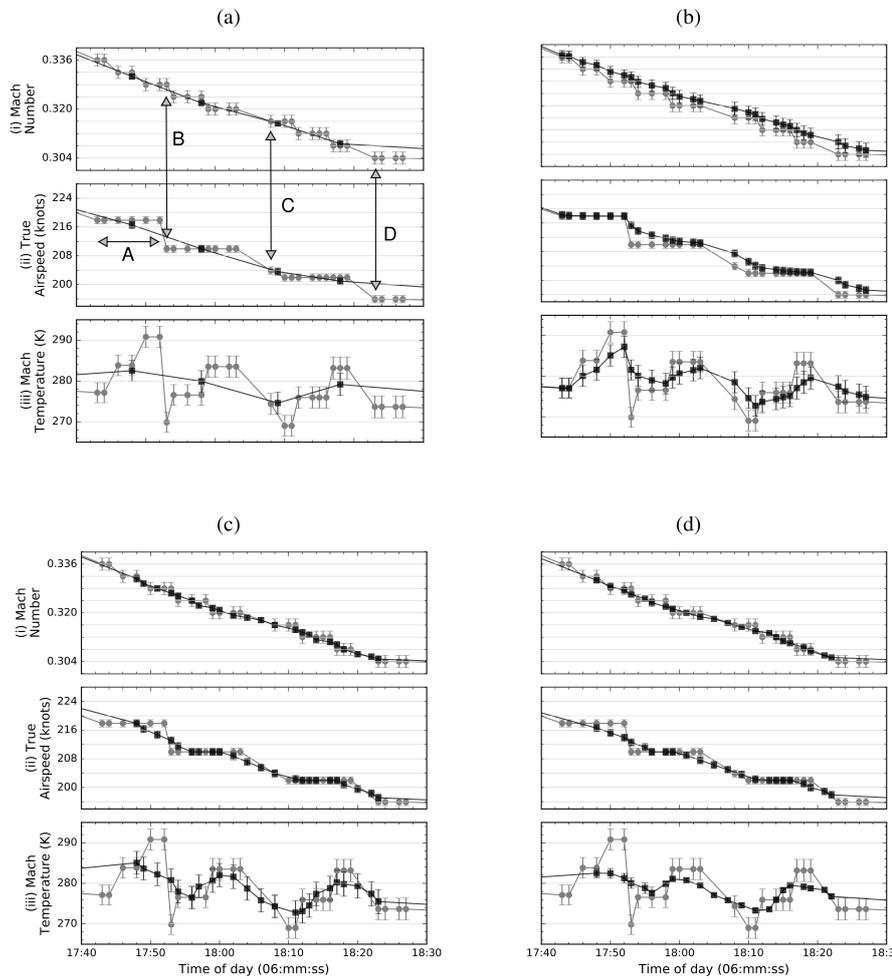


Figure 6: Before (circles) and after effects (squares) of applying smoothing filters for one aircraft's time series of (i) Mach number and (ii) true-air-speed for (a) Block Average, (b) Irregular Exponential, (c) Centred Moving Average and (d) Linear Regression. (iii) Mach Temperature computed before and after smoothing.

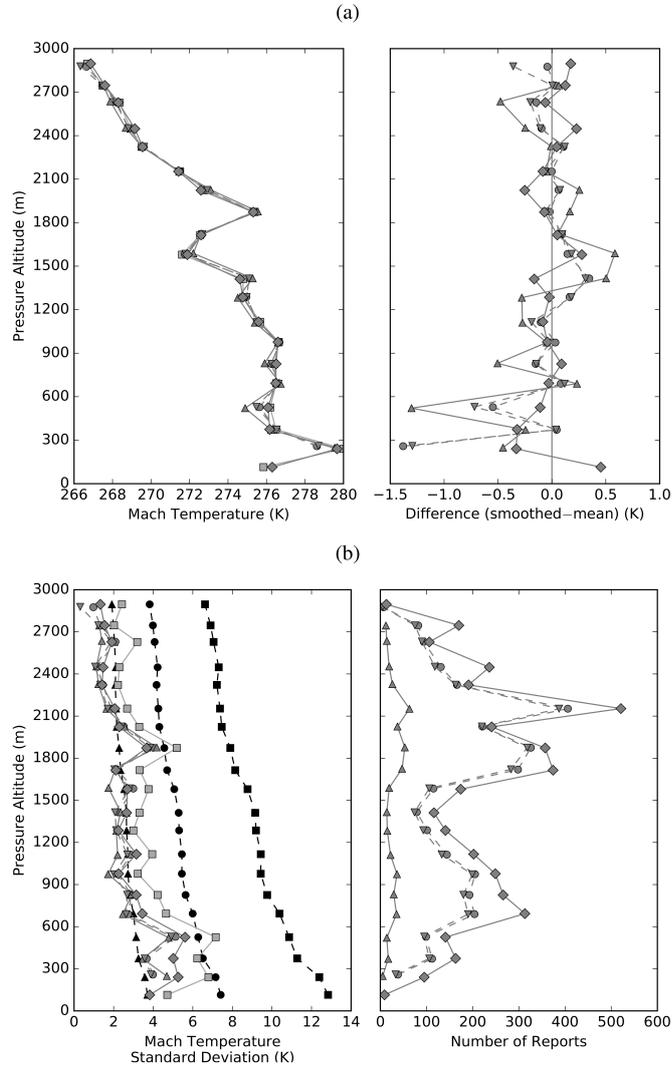


Figure 7: Effect of applying the different smoothing filters to Mode-S EHS reports of Mach number and true-airspeed along all aircraft tracks, London Heathrow domain, 4 January 2015 0530 to 0630. In each case (a) the resulting  $\bar{T}_{\text{MACH}}$  reports and (b) the estimated sample standard deviation are recomputed. Key:  $\blacksquare$  uncorrected  $\bar{T}_{\text{MACH}}$ , low-pass filtered:  $\blacktriangle$  BLK,  $\bullet$  CMA,  $\blacktriangledown$ , LIN,  $\blacklozenge$  IRR. Estimated error:  $\blacksquare$  full precision,  $\bullet$   $2 \times$  quantisation,  $\blacktriangle$  quantisation.

457 Figure 7(b)(i) shows the effect of each low-pass filter on the computed standard deviation of  
458 the  $\overline{T}_{\text{MACH}}$ . For comparison also shown are the expected standard deviations for the  $\overline{T}_{\text{MACH}}$ ,  
459 using the Mach Temperature error equation formulated by Mirza *et al.* (2016, Equation 16),  
460 assuming the following for the Mach number and airspeed: full precision error, precision due  
461 to quantisation error (Mirza *et al.*, 2016, figures 4 and 11) and precision due to double the  
462 quantisation error.

463 We used four low-pass filters: centred moving average (CMA), block average (BA), linear  
464 regression (LR) and irregular exponential smoothing (IRR). For smoothing the time series of  
465 reports above an altitude of 1000 m, the performance of each of the low-pass-filters was similar.  
466 Below 1000 m there was a small difference between using the moving window methods and  
467 the IRR. The former methods reduce variance more than the IRR. However, the advantage of  
468 the IRR method is that it uses all the available reports whereas the moving window methods  
469 removed reports as a result of the imposed quality control criterion. Furthermore, the IRR's  
470 weighting function is time-dependent, giving most weight to the most recent datum. This may  
471 reduce over-damping of high-frequency signals in the presence of a temperature inversion that  
472 would otherwise be smoothed by the moving window methods. However, each of the methods  
473 used to minimise the fluctuations in the Mode-S EHS derived observations, i.e., aggregation  
474 and low-pass filtering, effectively reduce the space and time resolution of the data.

## 475 7 Summary and Conclusions

476 This paper used Mode-S EHS reports exchanged between an aircraft and air traffic control to  
477 derive Mach Temperature. Using an aggregation of Mach Temperature reports from all air-  
478 craft within a defined region of an airport, e.g., the London Heathrow domain, vertical profiles  
479 of the mean Mach Temperatures,  $\overline{T}_{\text{MACH}}$ , for horizontal layers were constructed and used to  
480 identify a meteorological feature, temperature inversion, which is important for operational  
481 aviation weather forecasting and numerical weather prediction. To improve the representation  
482 of  $\overline{T}_{\text{MACH}}$ , low-pass filters were applied to the time series of Mode-S EHS reports of Mach  
483 number and airspeed for all aircraft within the London Heathrow domain. The low-pass fil-  
484 ter smoothed the discrete transitions of the Mach number and airspeed, which occur due to  
485 their low precision. Anomalous values of the derived Mach Temperature, which arise due to  
486 the asynchronous change between the Mach number and airspeed, were also smoothed. The  
487 overall effect of the low-pass filter reduced the variance of the  $\overline{T}_{\text{MACH}}$  by as much as 50%.

488 We compared hourly  $\overline{T}_{\text{MACH}}$  profiles with in situ observations of temperature reported by ra-  
489 diosonde and AMDAR, when available. We found that the  $\overline{T}_{\text{MACH}}$  profile between 1000 m and

490 3000 m shows some agreement with these in situ observations whereas below 1000 m there  
491 was less agreement, where the magnitude of the difference between the in situ observations  
492 and the  $\overline{T}_{\text{MACH}}$  was as great as 6 K. In our comparisons (figures 2, 3,4 and S3),  $\overline{T}_{\text{MACH}}$  seems  
493 to be in reasonable agreement with AMDAR and radiosonde data down to 600-700 m, a little  
494 lower than the 1000 m limit that we conservatively estimated. However, the results also show  
495 that some significant deviations can occur between 600 m and 1000 m. These arise in the early  
496 morning and the late evening, when there are few aircraft and so fewer Mode-S EHS reports  
497 at the lower levels. This scarcity may be due to the interruption of the line of sight between  
498 the aircraft and the Mode-S EHS receiver station. Hence we chose 1000 m as a safe lower  
499 limit for practical application. Daily operations may achieve better but this is best left to the  
500 meteorologist's judgement as they gain experience with the application.

501 However, the comparison against in situ observations is difficult since these are point based  
502 values, measured on time-scales of seconds to minutes, compared with the hourly mean of the  
503 aggregated Mach Temperature. Moreover, the radiosonde observations are not located within  
504 the airport domains. The temperature differences observed below 1000 m are unlikely to be  
505 due to changes in the ambient temperature; nor the prevailing meteorological conditions at the  
506 surface on the day (near freezing conditions, low wind speed and fog) but more likely due to  
507 Mode-S EHS processing (de Haan, 2011; Mirza *et al.*, 2016; Mirza, 2017; Stone, 2017).

508 We also compared the hourly aggregated Mach Temperature against the UKV model forecasts.  
509 We found similar results to our comparison with in situ observations. Furthermore, we found  
510 that the Mach Temperature profiles identified regions where temperature inversions may be  
511 present but which were not present in the UKV forecast, thus showing that Mach Temperature  
512 profiles may provide additional information for use in NWP.

513 From analysing the time series of the Mode-S EHS reports, we found that the Mode-S EHS  
514 processing also results in step changes in the reports of Mach number and airspeed that are  
515 asynchronous in time. This results in very large fluctuations in the corresponding Mach Tem-  
516 perature, ranging from 5 K to 9 K between adjacent reports.

517 We conclude that applying a low-pass filter to the time series reports of Mach number and air-  
518 speed could be beneficial as a pre-processing step prior to NWP data assimilation but further  
519 research would be needed in order to tune the filter parameters. Moreover, the IRR method  
520 could be used as the basis for a Kalman filter. While the quantitative value of the mean Mach  
521 Temperature may have a large uncertainty, the qualitative value of the constructed vertical pro-  
522 file of the mean Mach Temperature may provide additional information that may be useful for  
523 operational meteorology, e.g., identifying the possible locations for the occurrence of tempera-

524 ture inversions, when combined with other available sources of information. Furthermore, this  
525 may help aviation meteorologists to improve their forecasts for ATM by verifying in near-real-  
526 time the performance of the NWP forecast. However, further studies should be undertaken to  
527 assess this aspect.

528 The most common Mode-S EHS report is the aircraft's state vector from which temperature  
529 and horizontal wind observations can be derived. However, an alternative to Mode-S EHS is  
530 Mode-S MRAR (Strajnar, 2012; Strajnar *et al.*, 2015), but the current regulatory environment  
531 does not require aircraft or ATM to make such reports available. The technology and capability  
532 already exist for the direct reporting by aircraft of the temperature and horizontal wind. There-  
533 fore, in the interest of making more effective use of aircraft based observations for operational  
534 meteorology and numerical weather prediction, the aviation industry should be encouraged to  
535 implement either Mode-S MRAR reporting or its planned successor ADS-B.

## 536 **Acknowledgement**

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543 Edmund K. Stone (ed.stone@metoffice.gov.uk), subject to licensing conditions.

## 544 **Supplementary Section.**

545 In this supplementary section, in figures 8 and 9 we show for comparison the spatial distribution  
546 of Mode-S EHS reports for London Heathrow and London Gatwick, received between 1200 to  
547 1300 on 4 January 2015. London Gatwick is located 40 km south east of London Heathrow  
548 airport. At this time the air traffic flow was east to west, with aircraft arriving from the east  
549 and departing to the west. Each domain extends for a distance of 80 km east-west, 40 km  
550 north-south, height 3000 m from the surface, with the airport at the domain's centre. Points  
551 where the aircraft's roll angle is greater than  $5^\circ$ , i.e. when turning, are removed since these  
552 data are considered unreliable. While the domains appear to be cuboid this is not the case. The  
553 sampled volume of space resembles an inverted truncated pyramid. Figures 10 and 11 we show  
554 the vertical temperature profile for the London Gatwick domain for two separate time periods.  
555 The method used to compute the  $\bar{T}_{\text{MACH}}$  observations is described in section 5.1.

556 In figure 10, as noted in section 5.1, the Herstmonceux (45 km south east of Gatwick) ra-  
557 diosonde temperature profile (black line) shows that temperature inversions are present. The  
558 UKV temperature profile forecasts a low-level temperature inversion between 150 m and 300 m.  
559 The  $\bar{T}_{\text{MACH}}$  observations suggest that the upper-level inversion is at 1600 m rather than around  
560 2000 m shown by the radiosonde. Furthermore, the  $\bar{T}_{\text{MACH}}$  observations suggest that there  
561 is an isothermal region between 800 m and 1600 m, which is not shown by the UKV fore-  
562 cast or radiosonde. We note that there were no AMDAR reports for this period and location.  
563 The  $\bar{T}_{\text{MACH}}$  observations suggest that the rate of decay of the temperature inversion was much  
564 slower than that shown by the UKV forecast.

565 In figure 11, as noted in section 5.1, the Herstmonceux radiosonde temperature profile (black  
566 line) shows that temperature inversions are present. The UKV forecasts similar temperature  
567 inversions, although lower down when compared with the radiosonde. For this period and  
568 location there were five AMDAR reports, however, these do not show clearly the location of  
569 the temperature inversions. The  $\bar{T}_{\text{MACH}}$  observations show clearly the presence of the upper-  
570 level inversion but suggest it is lower down than forecast.

571 In both these cases, the  $\bar{T}_{\text{MACH}}$  observations at low levels may not be reliable because of the low  
572 number of Mode-S EHS reports used to make these report, as indicated by the width of the 95%  
573 confidence intervals, and the general increase in error at levels below 1000 m. Nonetheless, the  
574  $\bar{T}_{\text{MACH}}$  observations may provide useful information when compared alongside other in situ  
575 temperature observations.

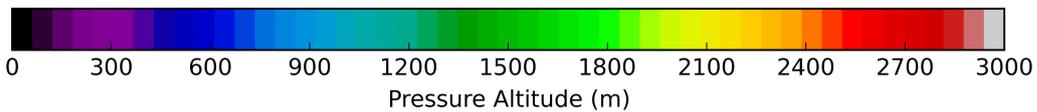
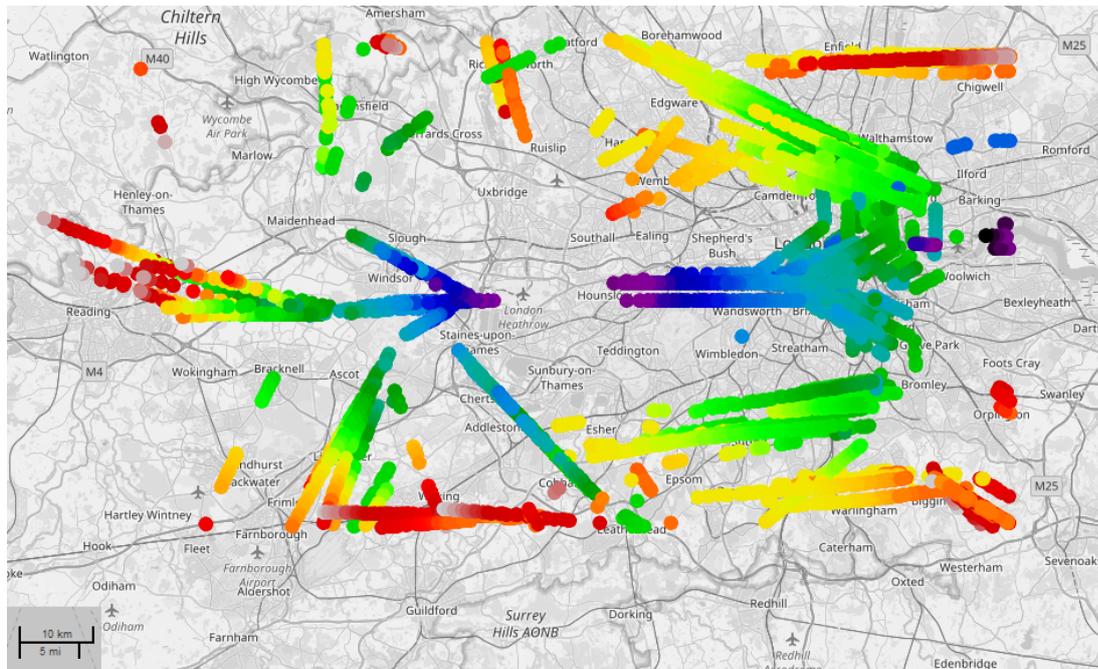


Figure 8: Spatial distribution of trajectories (circles, colour-coded by altitude) for ascending and descending aircraft within the London Heathrow domain, derived from Mode-S EHS reports received between 1200 to 1300 on 4 January 2015. (Cartography ©OpenStreetMap contributors, licensed as CC BY-SA <https://www.openstreetmap.org/copyright>, 2018)

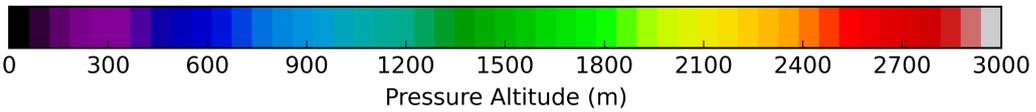
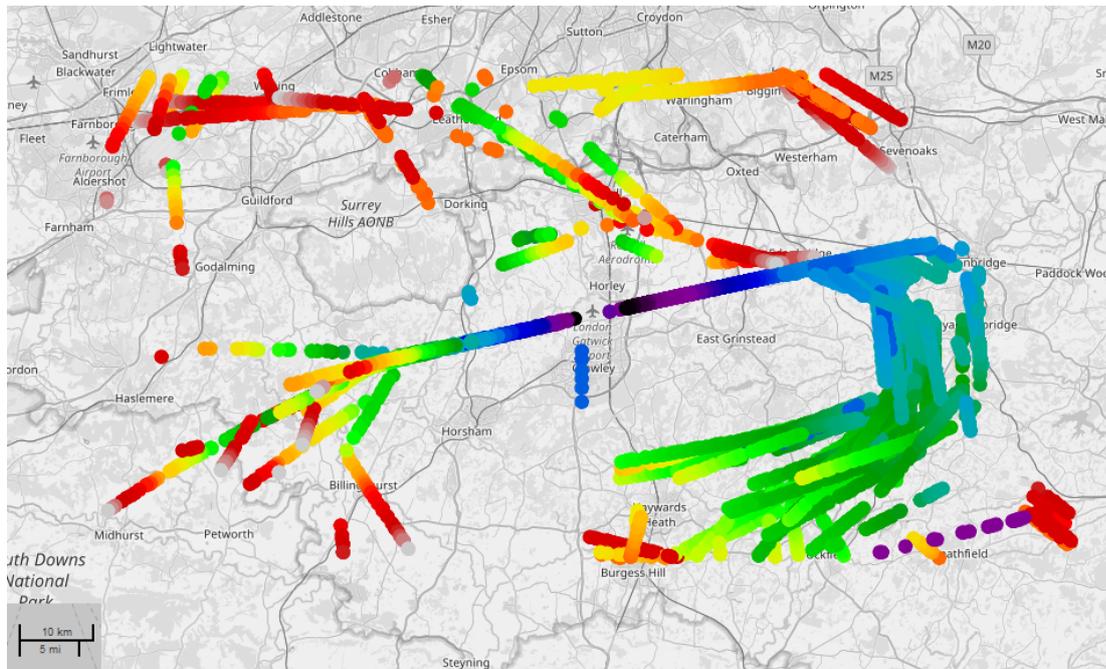


Figure 9: Spatial distribution of trajectories (circles, colour-coded by altitude) for ascending and descending aircraft within the London Gatwick domain, derived from Mode-S EHS reports received between 1200 to 1300 on 4 January 2015. (Cartography ©OpenStreetMap contributors, licensed as CC BY-SA <https://www.openstreetmap.org/copyright>, 2018)

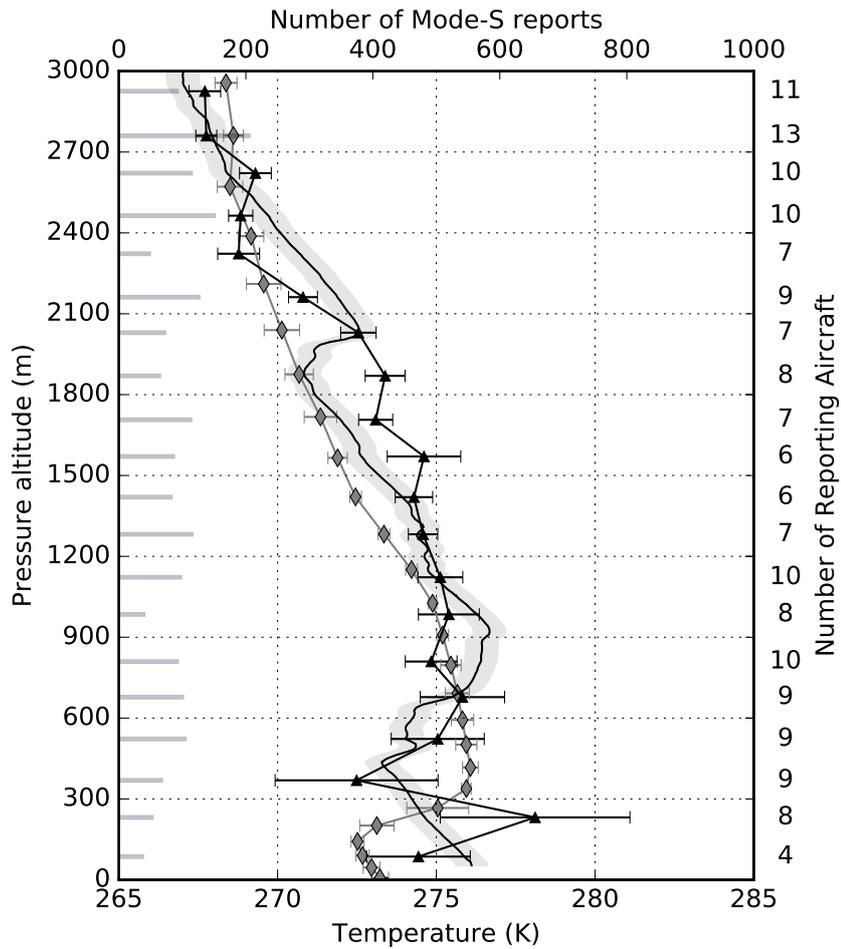


Figure 10: London Gatwick 2015-01-04, Mode-S EHS aggregated Mach Temperature vertical profiles (triangles), radiosonde (black) and mean UKV (narrow diamonds) temperature profiles. Symbols are as described in figure 2.

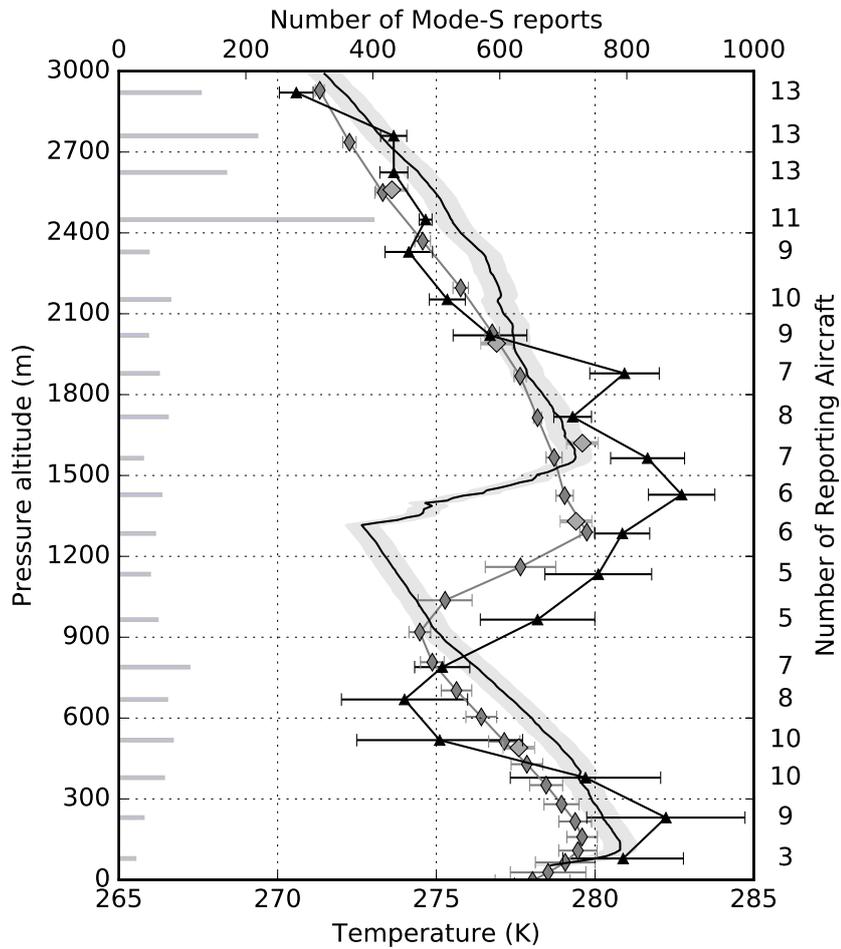


Figure 11: London Gatwick 2015-01-05, Mode-S EHS aggregated Mach Temperature vertical profiles (black triangles), radiosonde (black), available AMDAR reports (grey triangles) and mean UKV (narrow diamonds) temperature profiles. Symbols are as described in figure 2.

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