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Measurements and computations of flow in an urban street system

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8 Abstract We present results from laboratory and computational experiments on the

⁹ turbulent flow over an array of rectangular blocks modelling a typical, asymmetric

¹⁰ urban canopy at various orientations to the approach flow. The work forms part of a

¹¹ larger study on dispersion within such arrays (project DIPLOS) and concentrates on

¹² the nature of the mean flow and turbulence fields within the canopy region, recognis-

¹³ ing that unless the flow field is adequately represented in computational models there

is no reason to expect realistic simulations of the nature of the dispersion of pollutants
 emitted within the canopy. Comparisons between the experimental data and those ob-

tained from both large-eddy simulation (LES) and direct numerical simulation (DNS)

are shown and it is concluded that careful use of LES can produce generally excellent

agreement with laboratory and DNS results, lending further confidence in the use of

¹⁹ LES for such situations. Various crucial issues are discussed and advice offered to

²⁰ both experimentalists and those seeking to compute canopy flows with turbulence

²¹ resolving models.

22 Keywords Direct numerical simulation · Large-eddy simulation · Urban environ-

²³ ment · Wind-tunnel modelling

24 **1 Introduction**

²⁵ The use of large-eddy simulation (LES) to compute flow, turbulence and dispersion

²⁶ processes within urban environments is becoming ever more prevalent. This is partly

²⁷ because of continuously increasing computer power available to industry as well as

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in the academic environment, but also because of the recognition that lower order ap-28 proaches such as Reynolds-averaged Navier-Stokes (RANS) do not adequately cap-29 ture some of the important physics. Whilst LES has been common at larger scales 30 since Deardorff (1970) and, indeed, forms the basis of most large-scale numerical 31 weather forecasting models (in that processes on scales smaller than the grid are 32 parametrized), it has only within the last fifteen years or so been applied to the range 33 of much smaller scales and arguably greater complexities inherent in flow within ur-34 ban canopies. In such work, the urban canopy has normally been resolved (to varying 35 degrees of adequacy), rather than modelled in some way as is common in larger-scale 36 (mesoscale) computations. Initially, work concentrated on the flow field itself and 37 was generally aimed at computing cases that had been studied in the laboratory, (e.g. 38 Hanna et al., 2002; Kanda et al., 2004; Xie and Castro, 2006; Smolarkiewicz et al., 30 2007). More recently studies have included the assessment of scalar dispersion and 40 have also addressed specific field situations (e.g. Xie and Castro, 2009; Moonen et al., 41 2013). A useful recent review of the use of computational fluid dynamics for dis-42 persion in the urban environment has been provided by Tominaga and Stathopoulos 43 (2013), but the field continues to expand rapidly. (See also the review of Belcher et al., 44 2013). It is clear that model evaluation is important and this was addressed compre-45 hensively in the European COST action 732 programme (e.g. Schatzmann and Leitl, 46 2011). However, it is noteworthy that many such attempts (apart from COST732) 47 have concentrated largely on the adequacy of pollutant concentration results and not 48 on the underlying flow field. It is a truism to state that there is little reason to ex-49

⁵⁰ pect dispersion characteristics to be accurate if the underlying turbulent flow field is

⁵¹ inadequately predicted, unless there are counterbalancing errors of some kind.

In this paper attention is concentrated on (mostly) the canopy flow field for a neu-52 trally stratified boundary layer developing over an array of rectangular obstacles. Ex-53 periments in a large wind tunnel, in which the array is placed within a thick, simulated 54 atmospheric boundary layer (ABL), are reported and compared with correspond-55 ing LES data and also with fully resolved direct numerical simulations. The work 56 forms the first stage of a major project, DIPLOS (DIsPersion of LOcalised releases 57 in Street networks, www.diplos.org) whose objectives include generating greater un-58 derstanding of canopy flows so that rapid response modelling approaches based on 59 improved parametrizations can be developed for assessing the transport of potentially 60 hazardous releases in the urban environment. Reporting of the associated concentra-61 tion fields along with discussion of the extent to which current street-network models 62 adequately predict them will follow in a subsequent paper. Here we address both 63 the nature of the canopy flows for different wind directions and the extent to which 64 LES captures both the mean and the fluctuating flow, using comparisons between the 65 LES data and both laboratory and DNS data. The experimental and numerical ap-66 proaches are described in Sect.2. This is followed in Sect.3 by a discussion of the 67 upstream and above-canopy flows and then, in Sect.4 and Sect.5, by consideration of 68

⁶⁹ the within-canopy flow. Conclusions are summarized in Sect.6.

2

70 2 Methodologies

It has been traditional to use arrays of cubes (height h) in work of this kind be-71 cause this provides a geometry that leads to efficient DNS and LES computations (in 72 terms of the resources required). The typical case studied has a cube-to-cube spacing 73 74 equal to the cube size, which results in a rather open array compared with conditions in many city centres. The 'streets' between the intersections in such arrays are 75 only h in extent and this is inadequate for the establishment of the developed street-76 canyon flows that form the basis of street-network dispersion models (e.g. Soulhac 77 et al., 2011; Belcher et al., 2015) that are a focus of the current research. Ideally, 78 the street canyons should be long compared to h and of 1:1 or smaller aspect ra-79 tio (width:height). A compromise solution of $h \times 2h \times h$ blocks with h spacing was 80 adopted, acknowledging both these arguments and the implications in terms of com-81 puting resource. The latter consideration is all the more significant because an array 82 of at least 18 blocks was needed in the computations to attain results that were essen-83 tially independent of domain size. Note also that, despite its simplicity, the array is 84 a significant departure from the classical cube array in that it introduces geometrical 85 asymmetry and is thus more typical of real urban areas. 86 Nonetheless, there are many features of real urban areas that are not captured, e.g. 87 sloped roofs of different pitches on different buildings and non-parallel street config-88 urations. Although complex areas containing such features are occasionally modelled 89 in the laboratory and numerically (e.g. Yassin et al., 2005; Klein and Young, 2011, 90 as examples of specific city areas) and it is known that, for example, roof effects can 91 play an important role in dispersion, our eventual objective is to assess the adequacy 92 of street-network dispersion models and these are not yet available for more complex 93

situations. We can view the array used herein as a stepping stone between classical

⁹⁵ cube arrays and the more complex situations, but specifically chosen to allow eventual

⁹⁶ comparisons of dispersion behaviour with that predicted by existing network models.

97 2.1 Laboratory experiments

All experiments were conducted in the environmental wind tunnel in the EnFlo lab-98 oratory at the University of Surrey. This is an open circuit tunnel with a working 99 section that is 20 m long and 3.5×1.5 m in cross-section. The model canopy com-100 prised a square array of 294 $(14 \times 21) h \times 2h \times h$ rectangular blocks with height h = 70101 mm, mounted on a turntable whose axis of rotation was some 14 m downstream of 102 the test section entrance. The origin of the rectangular coordinate system was set at 103 the turntable (and model) centre, with x in the streamwise direction and z upwards. 104 Figure 1 shows the arrangement for the orientation defined as $\theta = 0^{\circ}$ – i.e. with the 105 oncoming flow perpendicular to the longer sides of the array obstacles. The array 106 was curtailed at its corners in order to fit the turntable (Fig.1b) and thus allow ease of 107 rotation to any desired angle. Note that the boundary layer upstream of the array was 108 initiated by a set of five Irwin spires, 1.26 m in height, and developed over surface 109 roughness comprising a staggered array of relatively sparsely distributed thin plates 110 $80 \text{ mm} \times 20 \text{ mm}$ (width and height, respectively), with spacing 240 mm in both x and 111



Fig. 1 (a) Plan view of the full array, showing coordinate notation and the domain size used for most of the LES and DNS (outlined in red). (b) Looking upstream in the wind tunnel. The array is in the $\theta = 0^{\circ}$ orientation with an Laser Dopplar Anemometer probe body visible above the array and the upstream spires that help to set the oncoming boundary layer just discernible in the distance.

y. The boundary layer at the start of the urban array (x = -2 m) was thus about 14h 112 in depth and was found to be reasonably homogeneous across the span with no sys-113 tematic spanwise variations. Measured velocities were within $\pm 5\%$ of the spanwise 114 mean. An internal boundary layer grew from the leading edge of the array, but con-115 ditions within the canopy, assessed for example by measurements along a spanwise 116 street for the $\theta = 0^{\circ}$ orientation, were essentially independent (i.e. within the experi-117 mental uncertainty) of the particular street downwind of the fifth street from the start 118 of the array. Two reference ultrasonic anemometers mounted downstream of the array 119 in the tunnel exit ducts were used to ensure that all the experiments were undertaken 120 at the same freestream velocity in the approach flow (2 m s^{-1}) . The Reynolds num-121 ber based on obstacle height and the velocity at that height in the upstream bound-122 ary layer was about 7,400, or about 830 when based on the friction velocity u_{τ} (i.e. 123 $Re_{\tau} = hu_{\tau}/v$, where v is the kinematic viscosity). The boundary layer was thus well 124 within the fully-rough-wall regime. 125

Velocity and turbulence measurements were made using a two-component Dan-126 tec Laser Dopplar Anemometer (LDA) system with a FibreFlow probe of outside 127 diameter 27 mm and focal length 160 mm. This provided a measuring volume with 128 a diameter of 0.074 mm and a length of 1.57 mm. Measurements in the local U - W129 plane within the street network (i.e. in planes aligned with the streets) were obtained 130 by use of a small mirror set at 45° beneath a downward pointing probe. The flow 131 was seeded with micron sized sugar particles at a sufficient level to attain data rates 132 around 150 Hz. In general, data collection times were 2.5 minutes, selected to control 133 the standard error in the results. This led to a typical standard error in U of 2%, in 134 $\overline{u^2}$ of 10% and in $\overline{w^2}$ of 5%, and corresponds to an averaging time of about 200T, 135 where T is defined as an eddy turnover time, $T = h/u_{\tau}$. Our confidence is based on 136 use of this LDA system over a long period of time, with a range or orientations and 137

geometries (with or without the mirror system). There were many instances of the 138 same variables being measured in different ways, without (for example) probe block-139 age problems becoming apparent. However, a potential source of significant error in 140 the measurements was due to positioning uncertainty relative to the local buildings 141 and tunnel co-ordinates. For example, an orientation error of 0.1° in the array align-142 ment to the wind tunnel axis would result in a positioning error of about 2.5 mm 143 relative to the buildings over a 1.5-m lateral traverse (i.e. in the y-direction), assum-144 ing the traverse itself to be perfectly aligned with the tunnel co-ordinates. There are 145 inevitable imperfections in any wind tunnel and traverse installation and these had 146 particular significance in this case because of the large volume over which results 147 were required. In broad terms, the positional error in any horizontal plane was typ-148 ically 2 mm. The implications obviously depend on the gradients of flow properties 140 at any given location and resulting uncertainties were greatest in the thin shear layers 150 downstream of the block surfaces (i.e. the side-walls and roof). The consequence of 151 small errors in height relative to the local building roof level were obvious in initial 152 experiments. This particular issue was resolved by use of a small ultrasonic height 153 gauge attached to the traversing arm – in this way local height uncertainties (i.e. rel-154 ative to the adjacent block) were reduced to about ± 0.5 mm. The results presented 155 here were obtained with this device in use (but see Sect.4 and Fig.11). 156

Further practical issues directly affecting the flow were the accuracy of rotation 157 of the array and its alignment relative to the approach flow. The 0^o orientation proved 158 by far the most demanding in these respects as any, albeit small, departure from the 159 ideal set-up generated a small cross-flow in the street network (see Sect.4). Dispersion 160 measurements would then show a plume axis that drifted to one side, as indeed was 161 observed in preliminary experiments that became the motivation for technique and 162 hardware improvements. Ultimately, these resulted in plume axis drift that was less 163 than 1°; it is hard to see that anything substantially better can be achieved. Finally, 164 it is worth noting that the 45° array orientation case was far less sensitive to these 165 matters, or rather that any consequent effects were far less obvious. 166

167 2.2 Salient LES details

The computations for array orientations of $\theta = 0^{\circ}$, 45° and 90° were undertaken us-168 ing the well-known OpenFOAM code, run on the University of Southampton's Iridis4 169 high performance computing system using typically 768 processor cores. Second-170 order differencing for the convective and diffusive terms was used everywhere and 171 time-stepping employed a second-order backward differencing scheme. Flow in a pla-172 nar channel whose domain size was $12h \times 12h \times 12h$ was simulated, although some 173 comparative cases were computed with smaller domain sizes (see Sect.4, where it 174 is shown that arrays much smaller were insufficient). The array of (smooth-walled) 175 obstacles was on the bottom (smooth) wall and comprised 24 obstacles – as shown 176 in Fig.1a – with no-slip conditions imposed on all surfaces, whereas at the top of the 177 domain stress-free boundary conditions were imposed. Periodicity was enforced in 178 the other two directions. All the statistics were obtained by averaging over at least 179 $\Delta T = 710T$, after an initial development period of at least $\Delta T = 420T$. Comments 180

about flow convergence will be made in due course. Whilst this approach to comput-181 ing rough-wall flows is common, we emphasise that the flow system is fundamentally 182 different to that in the wind tunnel where, as mentioned above, an internal boundary 183 layer develops over the array. However, the emphasis in this project is on the nature 184 of the flow and dispersion within the canopy rather than well above it. One of the 185 interesting questions we address in Sect.3 is the extent to which this canopy region 186 (below, say, z/h = 1.2) depends on the precise details of the outer boundary layer (or 187 channel) flow, at least for the range of outer flow conditions modelled in the labora-188 tory and by the numerics. It was anticipated that the dependence would not be very 189 significant and, indeed, this turned out to be the case. 190

A uniform mesh was used (providing formally better numerical accuracy than 191 more common expanding meshes) with a grid size of $\Delta = h/16$. Because the Reynolds 192 number was not very high ($Re_{\tau} \equiv u_{\tau}h/v \simeq 1000$) this was chosen to be near (but 193 above) the lower end of the range recommended by Xie and Castro (2006) for ade-194 quate simulation of urban areas and was a compromise driven by computer time lim-195 itations. The mixed time scale sub-grid model proposed by Inagaki et al. (2005) was 196 used; this circumvents either the (generally rather unsatisfactory) van Driest damp-197 ing function near the walls or the difficulties in removing the numerical instabili-198 ties which can arise near the walls if, to avoid using damping models, a dynamic 199 Smagorinsky model is implemented instead. These two difficulties can be particu-200 larly severe for cases (like the present) of multi-faceted wall geometry. However, 201 computations were also performed using the standard Smagorinsky model and only 202 small differences were observed in the spatially averaged mean velocities and turbu-203 lence stresses (less than 2% in mean velocities). Computations using smaller domain 204 heights (H = 6h, 8h or 10h) were also undertaken; some representative results will 205 be shown in Sect.4, confirming the weak effects of outer flow detail on canopy flow 206 statistics. The flow was maintained by enforcing a fixed axial mass flux. 207

208 2.3 Salient DNS details

Direct numerical simulations were carried out for the same building geometry at
orientations of 0° and 45°. The code was run on the UK national supercomputer,
ARCHER, using typically 240 cores. For detailed descriptions of the development of
the DNS code and the numerical techniques within it, see Yao et al. (2001), and for
examples of its use for urban boundary layer flows, see Coceal et al. (2006, 2007).

For the 0° case, the DNS was conducted in a somewhat smaller domain of size 12h × 9h × 8h, whereas the simulation of the 45° case was carried out in a domain of size $12h \times 12h \times 12h$ (as used for the LES). In both cases, a uniform grid resolution of $\Delta = h/32$ was used and the roughness Reynolds number achieved was $Re_{\tau} = 500$. This combination of mesh spacing and roughness Reynolds number was previously verified in similar studies to be adequate for a genuinely resolved DNS (e,g Coceal et al., 2006, 2007).

Periodic boundary conditions in horizontal directions were imposed. No-slip and impermeability conditions were prescribed at the bottom of the domain and on all solid surfaces, whereas free-slip boundary conditions were imposed at the domain's



Fig. 2 (a) Mean velocity profiles measured upstream of and over the array and; (b) the corresponding shear-stress profiles. Note that red symbols refer to the upstream boundary layer, blue symbols are profiles taken above the urban array. The vertical dashed lines in (b) indicate the estimated value of u_{τ}^2/U_e^2 in the two cases.

upper boundary. For both orientations, the flow was driven by a constant body force. 224 The flow Reynolds numbers based on the velocity at the top of the domain, U_e , and 225 the cube height, h, were typically about $Re_0 = 6,600$ and $Re_{45} = 7,500$ for the 0° 226 and 45° directions, respectively. By way of comparison, the corresponding Reynolds 227 numbers in the LES computations were in the range 14,500-16,000 and, in the wind 228 tunnel experiments, about 9, 300. 229 Both simulations were initially spun up until the turbulent flow was fully devel-230 oped, which was monitored by the convergence of statistical turbulence measures. 231

The time step for the simulations was set to $\Delta t = 0.00025T$ in both cases. Statistics were obtained from the converged simulations after a spin-up time of approximately ~ 210T (0°) and ~ 380T (45°), over averaging periods of $\Delta T_0 \simeq 650T$ and $\Delta T_{45} \simeq 320T$.

236 3 Results and initial discussion

237 3.1 The upstream boundary layer and its influence downstream

For reference purposes the major characteristics of the developed wind-tunnel bound-238 ary layer just upstream of the urban array are presented first. Figure 2a shows profiles 239 of axial mean velocity obtained just upstream of the array and also close to its cen-240 tre and within three streets of its downwind edge (x = -2000, -70 and 1190 mm, 241 respectively). Data have been spanwise averaged at each height, using the values 242 from various profiles taken at different spanwise locations. U is normalized by the 243 freestream velocity at each location. It is clear that there is very little boundary layer 244 growth over that fetch (although it is perhaps just noticeable by close inspection of 245 locations where $U/U_e = 0.95$, say). There is nonetheless a small increase in U_e with 246 fetch; normalizing by the tunnel reference velocity yields values of 1.013, 1.028 and 247 1.043 for the three locations. These changes imply a freestream acceleration param-248 eter defined by $(v/U_e^2)(dP/dx)$ of below 10⁻⁶, normally considered to have a negli-249



Fig. 3 Wind-tunnel profiles in the upstream boundary layer (near the front edge of the urban array). (a) Reynolds stresses normalized by u_{τ}^2 ; \bigcirc , $\overline{u'^2}^+$; \bigcirc , $\overline{v'^2}^+$; \bigcirc , $\overline{u'v'}^+$. Note that *h* here remains the urban array height, whereas the height of the upstream roughness elements is $h_u = 0.29h$. (b) Mean velocity data in logarithmic law form. The dashed line is the logarithmic law with d = 0, $z_o = 1.8$ mm ($z_o/h_u = 0.09$) and $\kappa = 0.41$.

gible effect on a regular turbulent boundary layer. The changes in U_e largely reflect the additional mass flux reduction in the inner part of the boundary layer over the array, evident in Fig.2a. The corresponding shear stress profiles are shown in Fig.2b, similarly normalized.

Note first that above a height of about 3h both the mean velocity and the shear 254 stress profiles at the downstream end of the array are very close to those upstream. 255 This suggests that the inner boundary layer growing as a result of the change of 256 surface condition does not reach beyond about z = 3h. Above that height, the flow 257 characteristics are essentially those of the upstream boundary layer. The immediate 258 implication is that the channel flow LES and DNS data might not be expected to 259 collapse onto the laboratory data above $z \approx 3h$. We return to this point in due course. 260 Spanwise-averaged centreline values of all the (non-zero) Reynolds stresses at 261 x = -2000 mm are plotted in Fig.3a, all normalized by u_{τ}^2 . The friction velocity, u_{τ} , 262 was estimated by assuming that the measured (spanwise-averaged) value of $-\overline{uw}$ in 263 the region just above the roughness is lower than u_{τ}^2 by a factor of 1.3, in accordance 264 with Cheng and Castro (2002) for a similar (but not identical) canopy morphology. 265 They showed that for arrays like these, this gave both a better match to the measured 266 form drag on the elements and a more satisfactory fit of the mean velocity data to the 267 logarithmic velocity law. In the near-wall region at least, the stresses are all typical for 268 a naturally grown boundary layer and, overall, they are similar to typical wind-tunnel 269 simulations of a neutrally stable atmospheric boundary layer. (Close inspection of the 270 outer region shows differences from a naturally grown layer, but these are immaterial 271 for the present purposes.) A measure of the adequacy of the estimated friction velocity 272 $(u_{\tau}/U_e = 0.067)$ is provided by Fig.3b, which shows the mean velocity plotted in 273 the usual logarithmic law form, $U^+ = \frac{1}{\kappa} \ln \left(\frac{z-d}{z_o} \right)$, and compared with the standard 274 logarithmic law assuming $\kappa = 0.41$. For the quite sparse roughness of this upstream 275 boundary layer, d = 0 provides a satisfactory fit even beyond what would normally be 276 expected as the logarithmic law range. This is an indication of the non-natural nature 277

of the outer flow. Note that the top of the roughness is at $z/z_o \approx 11$; depending on the

precise location of the measurement point in the x - y plane one would not necessarily

expect the logarithmic law to be followed much below $z/h_u = 2 (z/z_o = 22)$, where h_u

is the height of the roughness (20 mm), since such heights would be in the roughness

sublayer region where the flow must be inhomogeneous in both x and y.

As noted earlier, over the urban canopy an inner boundary layer grows and we

expect significant changes in the friction velocity and the two logarithmic law param-

eters d and z_o after the upstream edge of the array. This is explored in the following

section, where comparisons with the LES and DNS data are included.

²⁸⁷ 3.2 Flow above the urban array

The major focus within the DIPLOS project is the canopy region itself (i.e. flow, 288 turbulence and dispersion in and just above the $z \leq h$ region) but it is of interest 289 first to consider the flows above the canopy and for various wind directions. Figure 290 4 presents mean velocity and shear stress profiles for array orientations of $\theta = 0^{\circ}$, 291 45° and 90°, comparing laboratory, LES and DNS data. The computed profiles have 292 been obtained by averaging not only in time but also over the entire computational 293 domain. They are therefore not expected necessarily to agree with the laboratory data 294 in the roughness sublayer region (were the flow is homogeneous in neither x nor y), 295 since the latter data were obtained at specific x, y locations. Although the plan area 296 density is $\lambda_p = \frac{1}{3}$ independent of wind direction (with λ_p defined in the usual way 297 by the ratio of the plan area of the elements to the total plan area), intuitively one 298 would expect the surface drag for the zero degree case to be higher than for the 90° 299 case. The frontal area density (λ_f , the ratio of the element frontal area 'seen' by 300 the oncoming flow to the total plan area for a repeating unit) is $\frac{1}{3}$ for $\theta = 0^{\circ}$, i.e. 301 twice that for $\theta = 90^{\circ}$, so the former orientation provides a greater flow 'blockage'. 302 This larger drag for $\theta = 0^{\circ}$ is immediately evident: just above the canopy both the 303 measured and the computed shear stress for $\theta = 90^{\circ}$ are significantly higher and the 304 computed mean velocity profile shows a greater velocity deficit. The largest drag, 305 however, occurs in the $\theta = 45^{\circ}$ case, for which the near-wall shear stress reaches 306 values some 13% higher than the 0° values. This is consistent with a slightly higher 307 λ_f (0.35, cf. 0.33 for 0°) but perhaps more importantly with the fact that there are no 308 continuous streets in the prevailing wind direction for this particular orientation of 309 the array. 310

The flow parameters are normalized using the freestream velocity (or the velocity at the top of the domain in the LES and DNS cases), so do not collapse across the three orientations. Normalizing using the appropriate friction velocity leads to the corresponding profiles in Fig.5, from which it is evident that computational data in the inner region are in as good agreement with experiment as can be expected, especially given the uncertainty in establishing the friction velocity for the laboratory profiles (discussed above).

Note, first, that above the canopy neither the LES nor the DNS stress profiles (Fig.5b) collapse exactly onto the expected straight line between (0, 12) and (1, 0).

 $_{320}$ (12*h* = 840 mm, the domain height). However, they do collapse when the dispersive



Fig. 4 Mean velocity profiles (a) and shear-stress profiles (b) for the three urban array orientations. Note the location of the top of the canopy, shown as a dashed line at z/h = 1 in (b).



Fig. 5 Data of Fig.4 normalized using wall units. In (b), the dashed straight line joins the points (12,0) and (1,0).

shear stresses are added in (not shown) and it was the slope of these total stress lines in 321 un-normalized form that provided the LES wall stress values. (In these computations 322 the OpenFoam code was set to maintain a constant mass flux at each time step so, 323 without time-averaging the computed pressure difference across the two ends of the 324 channel, this was the most straightforward way to deduce the effectively imposed but 325 initially unknown wall stress. In the DNS, the known u_{τ} was forced by the applied, 326 constant pressure gradient.) The fact that the dispersive stresses (particularly in the 327 0° and 90° cases) were not exactly zero above, say, z/h = 2 could be a result either 328 of insufficient time averaging or, more likely, the presence of axial rollers in the outer 329 flow which, as a result of the rather small span, could not move around much in 330 the spanwise direction. It is interesting, however, that in the 45° case the dispersive 331 stresses above the canopy were closely zero. The effective span of the domain actually 332 varies with x in this case and it may be that this (and the effectively variable domain 333

Case	u^*/U_e	u^*/U_{2h}	Jackson d/h	z_o/h	κ
LAB $\theta = 0^{\circ}$	0.0748	0.119	0.62	0.086	0.33
LAB $\theta = 45^{\circ}$	0.0891	0.142	0.59	0.039	0.39
LAB $\theta = 90^{\circ}$	0.0557	0.078	0.64	0.053	0.265
*LAB $\theta = 90^{\circ}$	0.0557	0.078	0.86	0.009	0.39
LES $\theta = 0^{\circ}$	0.0678	0.123	0.62	0.080	0.33
LES $\theta = 45^{\circ}$	0.071	0.134	0.59	0.077	0.39
DNS $\theta = 45^{\circ}$	0.067	0.132	0.62	0.082	0.37
LES $\theta = 90^{\circ}$	0.0550	0.0863	0.64	0.064	0.265

Table 1 Parameter values deduced from laboratory and LES and DNS data. Note that all values for d/h were derived from LES or DNS results, except in the fourth line marked by an asterisk. There, $\kappa = 0.39$ was chosen and *d* varied to produce the best fit.

³³⁴ length across the span) prevents altogether the appearance of essentially fixed outer ³³⁵ layer axial structures. Incidentally, it is worth emphasising that the issue of domain

width for channel flow computations and whether or not it is sufficient to allow the

possible presence of axial rollers in the outer flow is also important for smooth-wall

flows (Fishpool et al., 2009).

Secondly, note that the only DNS data obtained with the H = 12h domain height 330 were for the $\theta = 45^{\circ}$ case and these data suggest a somewhat lower surface drag, 340 yielding a higher value of U_e/u_{τ} , most evident in Fig.5a. The LES and DNS profiles 341 in Fig.4a collapse quite well, but the corresponding collapse seen in Fig.4b required 342 the 6% higher value of U_e/u_{τ} (implied by Fig.5a) for the DNS case. This could be a 343 result of slight inadequacies in the subgrid model used in the LES but it could also be 344 partly explained by the difference in Re_{τ} , with the DNS value of 500 being about one 345 half that used for the LES. The issue is not important for the present purposes, given 346 our focus on flow variables (normalized by u_{τ}) in the canopy region, but it will be 347 fully explored in a subsequent paper in which results from computations using various 348 subgrid models and Reynolds numbers will be compared with the fully resolved DNS 349 data. 350

Thirdly, it is seen that for the 90° case the LES and laboratory mean velocity and 351 shear stress profiles agree quite well over much of the domain. In this case the obsta-352 cle array in the wind tunnel provides the least perturbation to the upstream boundary 353 layer. There is a much more significant perturbation in the other two cases, so the 354 wind tunnel profiles over the centre of the array consist more obviously of an inner 355 region in equilibrium with the new surface and whose depth grows with fetch over 356 the array, and an outer region which reflects the characteristics of the upstream sur-357 face. The friction velocity consistent with the inner region (increasingly large in the 358 sequence 90°, 0°, 45° for a fixed U_e) is thus appropriate for collapsing the LES and 359 laboratory data only in this inner region, consistent with the behaviour shown in the 360 figure. 361

Fourthly, as explained in Sect.3.1, the laboratory friction velocities were estimated by increasing the shear stresses obtained just above the canopy by the factor 1.3, in accordance with the findings of Cheng and Castro (2002). Table 1 lists the wall stresses for all three orientations, along with corresponding best-fit log-law parameters, which are discussed next. For the fits, the zero-plane displacement height, *d* was



Fig. 6 Mean velocity profiles in log-law form. The logarithmic law parameters $(d/h, z_o/h \text{ and } \kappa)$ are given in Table 1. In (b) 'Upper set' data refer to those from a probe traverse largely above the canopy height, whereas 'Lower set' data are from a separate traverse concentrating on the canopy region only.

assumed to be the height at which the surface drag appears to act (Jackson (1981)) 367 and was calculated from the LES and DNS data using the computed pressure field on 368 the elements and the frictional forces on the surfaces. This leaves only κ and z_o , the 369 roughness height, as free parameters. The former was chosen to ensure a good match 370 for the slope in the U vs. $\frac{u*}{\kappa} \ln[(z-d)/z_o]$ plot and the latter was chosen to ensure 371 the correct amplitude. For the experimental data, a similar value of d was used but 372 slightly different values of z_o emerged (compared with those deduced from the LES 373 data). 374

It is worth noting here that the values of κ in Table 1 are often quite different 375 to the more classical value of 0.41, which was adequate for fitting the wind tun-376 nel's upstream boundary layer data. The Kármán measure defined by $z^+ \frac{dU^+}{dz^+}$ (where 377 $z^+ = z u_\tau / v$) was not always very closely constant over a reasonable range of z in 378 the computations; one expects a constant value of $1/\kappa$ for a significant logarithmic 379 law region. There is therefore some uncertainty in the estimate of z_o and, of course, 380 different values of κ make a direct link between the value of z_o/h and surface drag 381 for different cases problematic. A change in κ from 0.33 to 0.4, for example, typ-382 ically leads to about a factor of two change in z_o . Note too that there is no reason 383 to expect the 'universal' value of κ to emerge – 0.39 is a recent suggestion for this 384 by Marusic et al. (2013) – because the ratio δ/h is not really large enough to imply 385 adequate scale separation between inner and outer layers. An example of the changes 386 that occur if κ is fixed and d is allowed to vary is included in (the fourth line of) 387 Table 1 for the $\theta = 90^{\circ}$ case. Using the method described above this has the lowest 388 κ (0.265). However, fixing κ at 0.39 (for example) and adjusting d to give the best 389 fit to the experimental data requires a rather higher d/h and a very much smaller 390 z_o/h . This latter value is unrealistically small, but fixing d/h as the 'Jackson value' 391 yielded quite a poor fit and no region of constant Kármán measure (indeed, values 392 were quite far from the expected 1/0.39). We believe our method – given a known u_{τ} 393 and known d and adjusting κ to yield the correct logariothmic law slope – is the most 394 self-consistent. 395

³⁹⁶ Despite these inevitable uncertainties, there is reasonable agreement between the ³⁹⁷ laboratory and LES and DNS data and the resulting log-law profiles for each wind direction are shown in Fig.6. For consistency with the LES, the DNS log-law parameters used in Fig.6b were those used for the corresponding LES case. They differ slightly from the values (shown in Table 1) which produced the best fit to the Kármán measure.

As a final illustration of the boundary layer flow above the canopy, Figs.7a-c 402 shows the turbulence normal stress profiles for the $\theta = 0^{\circ}$ case. Comparisons for 403 the LES axial stress for different wind directions are shown in Fig.7d. Note first that 404 the experimental profiles of both axial and vertical stresses ($\overline{u'^2}^+$ and $\overline{w'^2}^+$), approx-405 imately collapse at different x locations, because they reflect the characteristics of 406 the upstream boundary layer. Only in the inner region would one expect significant 407 differences at different axial locations. Nonetheless, these is a hint that data in the re-408 gion $1 \le z/h \le 4$ at the downstream end of the array (x = 1190 mm) are a little higher 409 than further upstream. This is consistent with that downstream part of the flow being 410 more closely in equilibrium with the rougher surface, although it should be borne 411 in mind that stress profiles normalized by the friction velocity are very similar in 412 smooth-wall and rough-wall channels (Leonardi and Castro, 2010). It is notable that 413 the LES axial stress in the outer region (Fig.7a) is significantly larger than the exper-414 imental data whilst the differences in the other two components are smaller. This is 415 almost certainly because of the presence of a significantly non-zero dispersive axial 416 stress (not shown), suggesting either that the computation had not yet converged (in 417 time), or perhaps that there are residual large-scale motions in the outer flow, prob-418 ably as a result of the finite domain span, although if the latter were true one might 419 expect non-zero dispersive stresses in the other two stress components (and there 420 were none). Figure 7d shows that there seems to be a significant dependence on wind 421 direction in the axial stresses in the outer flow. The axial stress is noticeably lower 422 for the 45° wind direction; this is the case that has no residual dispersive stress in the 423 outer region. What is more significant is that the stresses within the canopy $(z/h \le 1)$ 424 are very strongly dependent on wind direction, as expected. It is to this canopy region 425 that we now turn. 426

427 **4 Flow within the canopy region**

Consideration of the flow field within the near-wall region begins by presenting, as 428 examples, the axial and vertical mean velocity ensemble-averaged profiles (for the 429 $\theta = 0^{\circ}$ case) for a location at the centre of the long street – defined as the street paral-430 lel to the longer sides of the array obstacles. In this section velocities oriented in the 431 street directions are used - so U_s , V_s are velocities normal and parallel, respectively, 432 to the long side of the obstacles. Only for $\theta = 0^{\circ}$ does $U_s = U$, $V_s = V$. There is very 433 good collapse between laboratory, LES and DNS profiles of U_s^+ obtained using the 434 12h domain length (Fig.8a), despite the different domain heights and widths used; 435 the agreement continues all the way to z = 6h and 8h (not shown). However, a profile 436 given by an LES run using a domain size significantly smaller in plan $(6h \times 6h)$ dif-437 fers from the others once z/h > 1. This must be a result of the narrower (and perhaps 438 also the shorter) domain used and the effect is further illustrated by the V_s^+ profiles 439 seen in Fig.8b. For this array orientation (0°) and symmetrical location of the pro-440



Fig. 7 Normalized stress profiles for $\theta = 0^{\circ}$. (a) axial; (b) spanwise; (c) vertical stresses. (d) Comparison of the LES axial stress for the three wind directions.



Fig. 8 Ensemble-averaged mean velocity profiles at the long-street centreline for $\theta = 0^{\circ}$; (a): U_s^+ , (b): V_s^+ .



Fig. 9 Ensemble-averaged mean velocity profiles in street coordinates at the centre of the street intersection for $\theta = 45^{\circ}$; (a): U_s^+ , (b): V_s^+ .

files with respect to the array blocks, one would anticipate a zero spanwise velocity at all heights. However, this is not found in either the experiments or the numerical computations and is indicative of a small, but definitely non-zero difference between the canopy and the domain-top mean velocity orientations. Note that the fact that the *V* profiles within the canopy in Fig.8b are all the same sign is in one sense a coincidence (whether the *y*-coordinate is at $+90^{\circ}$ or -90° to the *x*-direction in either the laboratory or the numerical domain is completely arbitrary).

Some limited tests in the laboratory showed that the unexpected non-zero V could 448 be removed by an appropriate rotation of the array (by only a degree or two). In the 449 numerical computations, the periodic conditions imposed at the spanwise extents of 450 the domain allow non-zero V and it appears that too small a domain width can pro-451 mote a spanwise flow through the entire domain height, leading to an effective (and 452 small) 'free-stream' flow angle at the domain top. By far the largest flow angle at the 453 top (about 1.3°) is given by the LES on the $6h \times 6h \times 6h$ domain and it appears that 454 this is sufficient to trigger much larger flow angles within the canopy - not dissimilar, 455 in fact, to the laboratory values (see Fig.8b). At z/h = 0.5, for example, this smaller 456 domain LES run yields a flow angle in excess of 45° relative to the sides of the ob-457 stacles (rather than the expected value of zero, but note that at that height the axial 458 velocity is very small). This whole issue emphasises the care that is required in un-459 dertaking either laboratory or numerical experiments for these types of canopies. The 460 reason for the non-zero spanwise flow at all heights in the computations is unclear; it 461 462 may be that the total drag (and thus energy expended) is lowest for a small non-zero flow angle and the computation naturally picks out this lowest-energy flow. Further 463 work would be needed before a definitive answer could be identified. It is possible 464 that the zero-degree case is somewhat pathological, as it is presumably relatively easy 465 for the flow to 'switch' intermittently to conditions either side of a strictly symmet-466 ric state. Imposing a small non-zero wind angle could thus arguably provide a more 467 satisfactory case for comparing wind tunnel and numerical models. 468

Similar examples of velocity profiles are shown in figure 9 for $\theta = 45^{\circ}$. Again, these are ensemble averaged across all corresponding street locations in the whole

domain. In this case, the LES and DNS results for U_s diverge for z/h > 1, consistent 471 with the plan-averaged profiles shown in Fig.6b and with a small difference in the 472 computed flow angles at the top of the domain (not shown). It is not clear why this 473 difference occurs. Because of the array asymmetry with respect to the flow at $\theta = 45^{\circ}$ 474 this topology is expected to yield a non-zero lateral force in a numerical channel flow 475 computation (i.e. a force at 90° to the drag force, defining the latter as the array force 476 in line with the flow direction at the top of the domain). As Claus et al. (2012) discuss, 477 such a non-zero force implies that the mean flow angle at the top of the domain must 478 be slightly inclined to the forcing direction. Our results are qualitatively consistent 479 with the earlier Claus et al. (2012) findings in that a non-zero angle shift occurs up to 480 some height above the array, although the deviation appears more pronounced in the 481 case of the LES (extending all the way to the top of the domain). 482

We turn now to profiles along the streets (rather than vertically through them), 483 focussing first on street centrelines near z/h = 0.5. Figure 10 shows some examples 484 of these and includes mean velocity (U_s) and the two major shear stresses along the 485 y street for the $\theta = 0^{\circ}$ array orientation (Figs.10a,c,e) and both mean velocities and 486 $\overline{u'_s v'_s}^+$ for the $\theta = 45^\circ$ orientation. As before, the computed data are ensemble aver-487 aged across all available parallel streets in the domain. Consider first the $\theta = 0^{\circ}$ case 488 (the left hand column of Fig.10). Note that the mean velocity shown (U^+ , Fig.10a) 489 is the velocity across the street, i.e. in the free-stream flow direction. So behind the 490 blocks the velocity is negative and relatively small, whereas between them it is pos-491 itive and much larger as the flow tends to sweep down the x streets in the main flow 492 direction. There is good agreement between the laboratory and computational data, 493 not just for this mean velocity (Fig.10a) but also for the Reynolds shear stresses 494 (Figs.10c,e). The fact that the local magnitudes of the $\overline{u'v'}^+$ stress (Fig.10e), which 495 on average across the span must be zero by symmetry, is about the same as those of 496 the other dominant stress (Fig.10c) is a clear indication of the very three-dimensional 497 and anisotropic nature of the turbulence field within the canopy. It is significant that 498 the domain height, which is different for all three computation profiles, again has no 499 significant effect on the canopy flow. 500

The level of agreement for the $\theta = 45^{\circ}$ case is not quite so good, although it is 501 interesting that the shear stress data shown in Fig.10f all collapse reasonably well. 502 On the other hand, whilst the computed LES and DNS mean velocities are satisfy-503 ingly close (Figs.10b,d, and all obtained with a 12h domain height), there is a rather 504 larger level of disagreement between them and the laboratory data. However, the latter 505 are quite scattered and clearly vary significantly depending which axial (x) location 506 was chosen for the traverse. For this array orientation the experiments to obtain data 507 within the canopy were particularly tricky, but special care was taken over the final 508 traverses at x/h = 1 (x = 70 mm), with data taken at much closer intervals in an at-509 tempt to identify the various peaks and troughs. These data are satisfyingly close to 510 the computed profiles. 511

Very accurate vertical positioning of the LDA probe is not crucial at z/h = 0.5, where the slopes in vertical profiles of the flow variables are not large. At z/h = 1, however, slopes *are* large (see Fig.8a for example) so that lateral profiles taken near this 'roof-top' position are subject to rather more uncertainty when compared with

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Fig. 10 Normalized velocity and stress profiles at z/h = 0.53 along streets for $\theta = 0^{\circ}$ (a,c,e) and $\theta = 45^{\circ}$ (b,d,f). Street coordinates are used throughout and the location of the array blocks is indicated at the bottom of each figure. In (a,c,e) the laboratory *y*-locations have, for convenience in comparison, been shifted by *9h* and in all plots the origin of coordinates in the numerical data files has been shifted to cover the lab range conveniently. Symbols refer to laboratory data. Filled black triangles (in b,d,f) are from more closely resolved traverses. The legend for (c,e) is that for (a) and the legend for (f) is that for (b).

⁵¹⁶ computed profiles. This is illustrated in Fig.11, which shows DNS lateral profiles ⁵¹⁷ of U^+ along the *y* streets at three mesh node points nearest z/h = 1, compared with ⁵¹⁸ laboratory data taken nominally at z/h = 1. It is clear that except near the peaks, most ⁵¹⁹ of the laboratory data points lie between the lateral DNS profiles at z/h = 0.984 and ⁵²⁰ 1.016, as expected. Although the mesh was coarser, LES results (not shown) are quite



Fig. 11 Lateral U^+ profiles at $\theta = 0^\circ$ along the *y* streets, near z/h = 1. Block locations are indicated at the bottom of the figure.



Fig. 12 Mean vertical velocity along the long (y) street centreline at z/h = 1.03 for the $\theta = 0^{\circ}$ case. The left-hand axes refer to both W^+ and W/U_{vec} whereas the right-hand axes refer to the flow angle, α , in the vertical plane. Block locations are shown at the bottom of the figures. (a) LES; (b) DNS.

similar. It is worth noting that the DNS profiles show small differences in successive sections of the array – for the z/h = 1.047 profile, for example, the peak U^+ around y/h = 9 is larger than at the equivalent locations around y/h = 6 and 3. This may suggest either incomplete statistical convergence or, more likely, it is the effect of essentially stationary longitudinal rollers above the array indicated by the non-zero dispersive stresses there, discussed in Sect.3.2.

527 5 Further results and discussion

Dispersion of pollutants within the canopy region depends partly on the extent to which the flow can transport material into or out of the canopy. Despite the important influences of turbulence, this will clearly depend somewhat on the nature of the mean vertical flow at the canopy top. Figure 12 shows the variation of mean verti-

cal normalized velocity (W^+) along the centreline of the y-streets (i.e. parallel to the 532 long faces of the obstacles) for the $\theta = 0^{\circ}$ case. Data were ensample averaged across 533 all available street centrelines in the domain and for the LES (Fig.12a) are at the 534 first mesh point height above z/h = 1 (z/h = 1.03) whereas, for the DNS (Fig.12b), 535 they are interpolated to the same height (from the data corresponding to the U^+ data shown in Fig.11). Data at the lower LES mesh point (z/h = 0.97) are similar to those 537 shown in Fig.12a. The figure includes variations of the ratio W/U_{vec} , where U_{vec} is 538 the magnitude of the velocity in the horizontal plane, and the angle to the horizontal 539 of the total mean flow vector. It is evident that there are regions of both inflow and 540 outflow - i.e. negative and positive W (as there must be when spatially averaged, but 541 not necessarily in individual profiles such as those at a specific x). The strength of 542 the mean flow is not particularly large, as seen by the variations of the flow angle 543 (in the vertical plane), which do not exceed about 5° at most. Similarly, although the DNS W^+ values differ noticeably from the LES (cf. Figs. 12a and 12b), they are small 545 compared with the horizontal component – the W/U_{vec} ratio is below 0.1 everywhere. 546 Perhaps the most interesting feature of Fig.12 is that over each repeating unit 547 (e.g. from y/h = 3 to y/h = 6) there is significant asymmetry in W^+ about the cen-548 tre (y/h = 4.5), independent of whether LES or DNS results are considered. This 549 is also evident in Fig.11. If the approach flow were at 90° to the block face and 550 the lateral side force on the canopy were zero, W should be symmetric about that 551 point. One must conclude that one or both of those requirements are not precisely 552 satisfied or, alternatively, that small numerical inaccuracies are sufficient to produce 553 this asymmetry. Unexpected asymmetry evidenced by non-zero lateral (V) velocities 554 was discussed in Sect.4 (in relation to Fig.8b) and it is perhaps not surprising that 555 this small asymmetry is most clearly seen within the separated shear layer around 556 z/h = 1 in quantities that have large gradients there and are anyway very small. The 557 computed flow angle at the top of the domain was only about 0.1° for this case and 558 the lateral array force (normal to the flow direction at the top of the domain and the 559 sum of pressure and viscous contributions) was also practically zero, as expected. 560 Note that the lateral force normal to the forcing direction must inevitably be zero in 561 a numerical computation, as explained by Claus et al. (2012). We therefore conclude 562 that small numerical inaccuracies are sufficient to produce the asymmetry in W and, 563 indeed, yield noticeable differences between the LES and DNS data in Fig.12 (there 564 were, likewise, differences between DNS and LES in the unexpected non-zero V val-565 ues within the canopy - Fig.8b). These differences might also be a result of small 566 differences in dispersive stresses just above the canopy. This all emphasises the point 567 that numerical computations of these kinds of flow are not as straightforward as one 568 might at first imagine - a salutary warning to computationalists! 569 Contour plots of W^+ at z/h = 1.0 are shown in Fig.13a for all array orientations. 570

⁵⁷⁰ Contour plots of W^+ at z/n = 1.0 are snown in Fig.13a for all array orientations. ⁵⁷¹ In every case, there are significant areas of outflow, as must inevitably be the case ⁵⁷² since the spatially-averaged mean value must be zero (at all heights, in fact, by mass ⁵⁷³ continuity). The regions of outflow, however, are different: for $\theta = 0^{\circ}$ they are con-⁵⁷⁴ centrated at the trailing edge of the obstacle roofs and downstream of the side edges ⁵⁷⁵ whereas, for $\theta = 90^{\circ}$, they lie along the side edges and front face. Since one might ⁵⁷⁶ intuitively have expected the obstacles to generate delta-wing type vortex motions ⁵⁷⁷ in the $\theta = 45^{\circ}$ case, it is interesting that there is, nonetheless, a region of outflow



Fig. 13 (a) Contour plots of the normalized mean vertical velocity, W^+ , at z/h = 1, from the LES data at all three array orientations. (b) For $\theta = 45^{\circ}$ and z/h = 0.5, contour plots of W^+ (left) and flow vectors (right) in the horizontal plane.

downstream of the rearmost corner. If the influence of turbulent fluxes at z/h = 1was negligible, these plots would indicate the regions where any pollutants emitted within the canopy would be expected to be transported out to the boundary layer above. Likewise, some would be transported back into the canopy from aloft in the regions of negative W^+ . However, it is likely that the effects of turbulent transport are equally if not more important; the issue will be explored in the subsequent dispersion paper, but it is worth noting here that Belcher et al. (2015) (for an array of cubical ob-



Fig. 14 (a) Tracers following the meanflow (i.e. mean flow pathlines) for the $\theta = 45^{\circ}$ case. The arrow shows the wind direction aloft. The right-hand sketch shows the origins of the nine coloured traces - equispaced in the street cross-section. The LES data were used. (b) Snapshot from video taken for $\theta = 45^{\circ}$. The ground-based square smoke source (70×70 mm), outlined by the white square, is located at the centre of a long street and the (green) laser sheet showing the smoke is coincident with the horizontal plane at z/h = 0.64 and is viewed from above.

stacles) suggest that, indeed, turbulent transport is dominant compared to advection
 with mean *W*, but this is probably not true near the upwind edge of the array or if the
 obstacle height varies significantly.

A similar contour plot is shown in Fig.13b for $\theta = 45^{\circ}$, but at the canopy half-588 height, z/h = 0.5. It is evident (see the left-hand plot) that the upward flows (positive 589 W^+) are considerably stronger and more extensive than those at the top of the canopy, 590 seen in Fig.13a (centre plot). To compensate, the downward flows, although restricted 591 to thinner regions near the edges of the blocks, have significantly greater magnitude. 592 The horizontal component of the total mean flow is shown in the vector plot (at the 593 right-hand side of Fig.13b). The recirculating region behind the rearward short faces 594 of the blocks can be seen, but the dominant feature is that the flow in the long streets 595 (parallel to the longer side faces) is predominantly in the along-street (y_s) direction, 596 despite the 45° wind direction aloft. This feature of canopy flows for wind directions 597 not normal to obstacle faces was discussed by Claus et al. (2012) and is likely to 598 remain a strong feature of urban canopies independently of the precise array geome-590 tries, unless the obstacle sizes and orientations are different from one another so the 600 array does not embody any long continuous streets. A similar 'street steering' effect 601 has also been observed in the field (e.g. Balogun et al., 2010; Carpentieri and Robins, 602 2010). Figure 13b suggests that the 2h streets of the present array are just long enough 603 to be representative for the street network modelling approach. 604

As an example of possible pollutant pathways in the absence of any turbulence 605 effects Fig.14a shows (from LES data) mean flow pathlines originating from a grid 606 of nine points in the vertical plane at the centre of the long (y_s) street and equally 607 spaced between themselves and the obstacle side walls. There is a helical flow within 608 the street but from some points the 'tracers' can escape above the canopy (via the 609 positive vertical mean flow regions discussed above) and then they rapidly align with 610 the mean flow aloft. Side views of the same results show that in no case do the tracers 611 reach heights above $z/h \approx 1.1$. 612

It is worth noting that data like those presented in Figs.13 and 14a would be 613 almost impossible to obtain from laboratory or field experiments. (An indication 614 of what can be achieved, however, is seen in Carpentieri et al., 2009). The fig-615 ures are therefore examples of the added value provided by numerical computa-616 tions and are clearly helpful in providing further understanding of the canopy flows. 617 They should be interpreted with care, however. As indicated earlier, the presence 618 of large-amplitude turbulent motions will ensure that tracers would not actually fol-619 low the mean flow particle paths shown in Fig.14a. We illustrate this by showing in 620 Fig.14b, for comparison with Fig.14a, a corresponding but instantaneous snapshot of 621 the smoke pattern arising in a laboratory experiment on a plane not far from the mid-622 height of the canopy. The source of smoke laden air was an area of size $h \times h$ at z = 0623 and located at the centre of a long (y_s -direction) street. It is clear that (i) some smoke 624 can move 'upstream' of the source location and (ii) some can arrive at considerable distances laterally within the canopy - much further than would be suggested by the 626 selected mean flow tracers of Fig.14a. The consequences of this rapid lateral spread 627 are sometimes seen in dispersion measurements in the field - for example, the mea-628 surements in central London described by Wood et al. (2009). Views of a horizontal 629 plane at z/h = 2 (not shown) indicate (iii) that the smoke can reach heights well in 630 excess of the z/h = 1.1 suggested by mean flow tracers and certainly above z/h = 2. 631 These three facts alone are sufficient to demonstrate that the turbulence fluxes are 632 very significant, so that mean flow tracers like those shown in Fig.14a should indeed 633 be interpreted with caution. It is crucial to study these fluxes in detail and this will be 634 a topic for the subsequent dispersion paper describing the concentration fields within 635 and above the canopy. 636 Not only are the turbulent fluxes important but it should be noted that, within the 637 canopy, dispersive fluxes – arising from the spatial variability of the local time-mean 638 velocities in horizontal planes – are also large. This is illustrated in Fig.15, using 630 the LES data. The data have been normalized in each case by the corresponding

Reynolds stress at the appropriate height and it is clear that they can be of the same 641 order as the latter over large parts of the canopy height, as found in previous studies 642 (e.g. Coceal et al., 2006). This emphasises the high degree of spatial variability of 643 flow properties within the canopy. Although in some circumstances pollutants may 644 be well mixed (so that concentrations are not too non-uniform) this does not imply 645 uniformity in the flow variables. Since the flows are strongly three-dimensional and 646 inhomogeneous within the canopy, the usual decomposition of stresses in coordinates 647 aligned with (e.g.) the forcing direction is perhaps not particularly useful; one could 648 argue that principle stress coordinates should be used. However, this seems an unnec-649 essary complication in the present context and would not add very much to physical 650

652 6 Final discussion and conclusions

understanding.

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⁶⁵³ We remark first on conclusions arising from the wind-tunnel experiments. Measure-

⁶⁵⁴ ments in an extensive array of this kind are particularly challenging, not least because ⁶⁵⁵ of the need to maintain positional accuracy relative to the array blocks whilst moving



Fig. 15 Vertical profiles of dispersive stresses within the canopy from the LES for $\theta = 0^{\circ}$ (a) and $\theta = 45^{\circ}$ (b). Each dispersive stress, at each height, is normalized by the corresponding (time- and domain-averaged) Reynolds stress at that height.

across several modules. The consequences are most obvious when traversing across 656 the shear layers in the flow separating from the building block roof and walls, as is 657 made very clear from inspection of the DNS results in Fig.11. Related issues arise 658 from the sensitivity of the flow to slight errors in alignment in the 0° and 90° cases. 659 Although considerable efforts were made to improve experimental techniques, these 660 matters remained the main cause of uncertainty in the data. The weak mean cross-661 flow seen in the computations for the 0° case implies a consistent, though weak drift 662 in the centre line of a plume dispersing through the array. Drift of this nature is likely 663 to be of greater magnitude in the wind-tunnel work, due to overall alignment error, 664 though variable to some degree, reflecting local errors in block alignment. These 665 matters will be returned to in comparing measured and predicted dispersion in the 666 subsequent paper. 667 Next, conclusions arising from the numerical computations are given. Firstly, it 668

has been shown that the computed flows within the present urban-type canopy are not
very sensitive to the domain height. This is significant, as it makes it computationally
more efficient to model pollutant releases within the canopy. Nonetheless, we recommend a domain height of at least six canopy heights in order to capture the most
important turbulence features just above the canopy, some of which are necessarily
linked to the turbulent flow at greater heights.

In common with previous work, some of our results suggest the possible pres-675 ence of longitudinal, slowly-evolving rolls above the canopy. These can be strongly 676 attenuated, if not completely damped out, if the computational domain is too small. 677 For the present canopy morphology, a domain plan area of $6h \times 6h$ seems too small 678 (see Sect.4), especially for flow directions normal to the obstacle faces; these direc-679 tions are in one sense pathological and allow the computed flow to break symmetry 680 and contain a mean spanwise flow that is increasingly enhanced as the domain size 681 decreases. The presence of slowly-moving rolls aloft also has implications for mod-682 elling limited-duration pollutant releases, because downstream concentration patterns 683 could depend somewhat on the location of the rolls (with respect to that of the source) 684 over the particular release and dispersion times. At this stage it is not clear how sensi-685



Fig. 16 Mid-height (z/h = 0.5) flow vectors for $\theta = 45^{\circ}$. (a) Square cube array - from Claus et al. (2012). (b) the present array; note that only half of each $h \times 2h \times h$ obstacle is shown, so that only downwind half of the obstacles is shown at the top of the figure and the upstream half at the bottom.

tive this feature is to the specific array morphology, but it is certainly something that should be considered in designing numerical experiments on such flows.

Secondly, as noted above, the present results illustrate the difficulty in achieving 688 perfect flow symmetry for cases where the geometry would lead one to expect it. This 689 is true both for laboratory and numerical modelling. It may be a result of the specific 690 canopy morphology having its lowest drag condition at some small angle to that for 691 which symmetry is expected, but further work would be needed to confirm this and, if 692 this is the cause, the behaviour would certainly vary with canopy morphology. What-693 ever the cause, this asymmetric feature is a further indication of the care needed in 694 designing and executing such experiments. In nearly all the extant literature, insuf-695 ficient data are shown to give confidence that such a spanwise (symmetry-breaking) 696 flow is *not* present, so the present results provide a further cautionary lesson. 697

Thirdly, the present canopy has obstacles sufficiently long compared with their 698 heights to yield extensive flow channelling along streets. This is most clearly illus-699 trated by Fig.16. The region in which the flow turns to become parallel to the long 700 sides of the obstacles is no more 1h in extent (in both x_s and y_s directions) – a lit-701 tle smaller than what was found in the more classical (square) cube array studies of 702 Claus et al. (2012), shown on the left of the figure. Across the whole of the down-703 wind half of the long street the flow for the present canopy is closely aligned with 704 the obstacle faces, despite the 45° flow orientation aloft. This supports the suggestion 705 made in Sect.5 that the streets are long enough to be representative for street network 706 modelling approaches; shorter streets would probably not be sufficient and it will be 707 interesting to see how well network models can predict concentrations in the present 708 canopy. That will be the subject of a forthcoming paper. 709

Finally, it is worth noting that the domain-averaged axial mean velocity profiles 710 through the canopy cannot be sensibly fitted by an exponential profile, for any of 711 the wind directions considered. MacDonald (2000) was perhaps the first to make 712 the suggestion that profiles could be so fitted (although such profiles in vegetation 713 canopies had long been proposed Cionco (1965)) and recently Yang et al. (2016) 714 have suggested that good fits to exponentials *can* be obtained for a wide range of 715 arrays comprising cubical obstacles. However, although they studied arrays of cubes 716 with $\lambda_p = 0.25$, identical to those studied by Coceal et al. (2006), Leonardi and Cas-717 tro (2010) and Claus et al. (2012), the canopy velocity profiles they obtained differed 718 significantly from those obtained by all these latter authors. It seems likely that their 719 mesh was not fine enough (having only eight points across the height of the canopy) 720 to resolve the thin shear layer at the canopy top. A 25% area coverage is almost within 721 the full 'skimming' regime ('d-type' roughness, in the classical roughness terminol-722 ogy) and it may well be that for much lower λ_p typical of 'k-type' roughness when 723 sheltering between obstacles is less prevalent, the velocity profiles can be reasonably 724 modelled by exponentials. This remains an open question which will be considered 725 in a further paper, but there is no doubt that the present computations can be used to 726 show that assumptions typically made to derive an analytical (exponential) velocity 727 profile model are generally far from valid in urban type canopies. 728

Despite the various uncertainties discussed in both the laboratory and the compu-720 tational studies, an important general conclusion of the work is that the computations, 730 whether by LES or DNS, satisfactorily capture the salient details of the complex, 731 three-dimensional flow within the canopy, in that the results agree as well as can be 732 expected with the wind-tunnel data. This is very encouraging, for it suggests that any 733 subsequent differences found between computed and laboratory statistics of disper-734 sion behaviour, for the same configurations and using the same methods, will not be 735 a result of inadequate flow computations. 736

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